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ATTENUATION, RIPPLE AND PHASE STABILITY  
IN VLA WAVEGUIDE TRANSMISSION SYSTEM

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In waveguide transmission line, there are many sources causing ripples in attenuation characteristics. Some of these affect the IF transmission characteristics and local oscillator system stability. These ripples are especially bothersome because they are affected by small waveguide length changes and hence are unstable.

In this report, all the ripple sources that can be considered are reviewed and stability against waveguide length change is calculated. The analysis will be done in the case of many ripple sources and the specification of couplers and some other components will be reviewed.

1. GENERAL RELATIONSHIP BETWEEN SPURIOUS SIGNAL LEVEL  
AND PHASE SHIFT

Under the following definitions:

- A ;  $TE_{01}$  mode amplitude (shown in vector)
- $A_0$  ;  $TE_{01}$  mode main signal amplitude (will be set to 1.0 in this report)
- C ;  $TE_{01}$  mode spurious signal amplitude
- $\phi$  ; phase shift caused by spurious signal
- $\psi$  ; phase difference between main and spurious signals

The amplitude and phase shift of TE<sub>01</sub> mode after addition of a spurious signal can be expressed as follows:

$$\begin{aligned}
 A &= |A|e^{j\phi} \\
 &= A_0 + Ce^{j\psi} \\
 &= 1 + Ce^{j\psi}
 \end{aligned}
 \tag{1-1}$$

When  $\psi$  changes with frequency, ripple is caused in  $|A|$  and its amplitude is given as:

$$\begin{aligned}
 R_{\max}^{\text{P-P}} &= 20 \log (1+C) - 20 \log (1-C) \\
 &\doteq 17.3C \text{ (dB) } (C \leq 0.1)
 \end{aligned}
 \tag{1-2}$$

Also the phase shift caused by a spurious signal can be expressed as follows:

$$\Delta\phi = \sin^{-1} (C\sin\psi)
 \tag{1-3}$$

and maximum peak-peak variation becomes;

$$\Delta\phi_{\max} = 2 \sin^{-1} (C).
 \tag{1-4}$$

Figure 1-1 shows the relationship between spurious signal level,  $R_{\max}^{\text{P-P}}$  and  $\Delta\phi_{\max}^{\text{P-P}}$ .

## 2. SPURIOUS SIGNAL SOURCES IN TE<sub>01</sub> MODE CIRCULAR WAVEGUIDE SYSTEM

The TE<sub>01</sub> mode spurious signals generated in waveguide can be classified into two kinds as follows:

- A. Spurious signal caused by mode conversion-reconversion between TE<sub>01</sub> mode and unwanted modes.
- B. Spurious signal caused by TE<sub>01</sub> mode reflections.

In both cases, one source of mode conversion or reflection does not cause ripple, but combinations or interactions of

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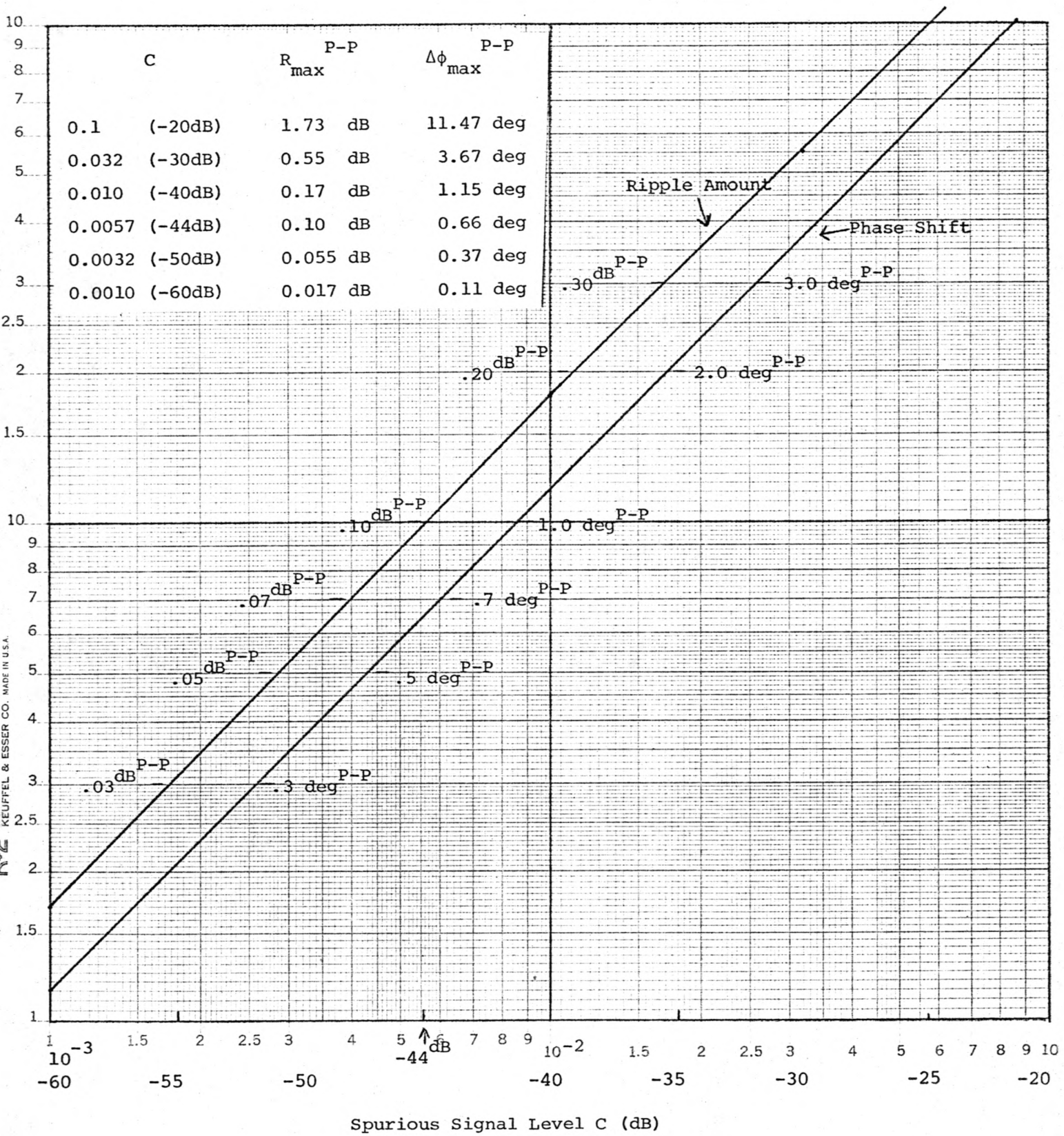


FIGURE 1-1: THE RELATIONSHIP BETWEEN SPURIOUS SIGNAL LEVEL AND PHASE STABILITY

more than two sources do cause ripple. Generally the ripple period depends on the distance between those sources, and the larger the distance becomes, the smaller the period becomes. For the VLA Waveguide System, it is important to suppress the ripples of less than 50 MHz period to a level of  $\leq 1$  dB peak-to-peak and ripples of less than 10 MHz period to  $\leq 0.2$  dB peak-to-peak.

### 2.1 Spurious Signal Caused by Mode Conversion Effect

Actual cases of the spurious signal caused by mode conversion effect can be listed as follows:

SOURCE	SPURIOUS MODES
A. Between taper waveguides	$TE_{on} (n \geq 2)$
B. Between flexible waveguides	$TE_{11} TE_{12} \cdots TE_{21} TE_{22} \cdots TM_{11}$
C. Between imperfect connections	$TE_{11} TE_{12} \cdots TE_{21} TE_{22} \cdots TM_{11}$
D. In uniform or random dimensional imperfections	Depends upon deformation
E. Between $TE_{on}$ mode generation source and helical coupler	$TE_{on} (n \geq 2)$ (especially $TE_{02}$ )
F. Between Michelson interferometer couplers (beam splitter couplers)	$TE_{on} (n \geq 2)$

Out of those, C and D are small enough to neglect when helix waveguides with random length (as NRAO Spec.) are used. Much attention must be paid to A, E, and F, because  $TE_{on} (n \geq 2)$  cannot be absorbed without special mode filter and the attenuation of  $TE_{on}$  mode in the main waveguide is too small to attenuate this effect.

### 2.2 Spurious Signal Caused by $TE_{01}$ Mode Reflection Effect

As  $TE_{01}$  mode reflection effects in the VLA waveguide system, the following can be listed:

1. Between couplers (in main line)

- A. Reflection from front port and back port
- B. Reflection from front port and reverse coupling
- 2. Between a coupler (front port) and signal distributor
- 3. Between a coupler (coupled port) and circular-rectangular adapter

### 3. MODE CONVERSION EFFECT

#### 3.1 Ripple Amount

When there are two mode conversion sources, as shown in Figure 3-1, the output  $TE_{01}$  amplitude is as follows:

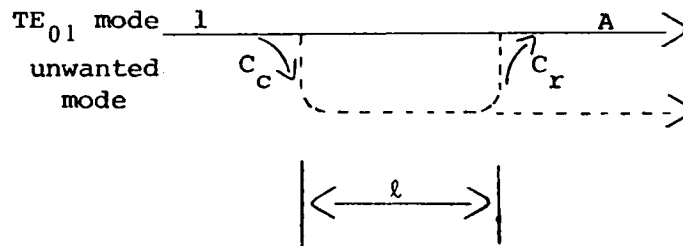


FIGURE 3-1

$$\begin{aligned}
 A &= |A|e^{j\phi} \\
 &= \sqrt{1-C_c^2} \cdot \sqrt{1-C_r^2} e^{-\alpha_{01}l - j\beta_{01}l} - C_c C_r e^{-\alpha x l - j\beta x l} \quad (3-1) \\
 &\doteq e^{-\alpha_{01}l - j\beta_{01}l} (1 - C_c C_r e^{-\Delta\alpha l} \cdot e^{-j\Delta\beta l})
 \end{aligned}$$

$C_c, C_r$ ; mode conversion coefficients

$\alpha_{01} + j\beta_{01}$ ; propagation constant of  $TE_{01}$  mode

$\alpha x + j\beta x$ ; propagation constant of spurious mode  
(unwanted)

$\Delta\alpha (= \alpha x - \alpha_{01})$ ; the difference in attenuation constant

$\Delta\beta (= \beta x - \beta_{01})$ ; the difference in phase constant

$l$ ; the distance between two mode generation sources

In this case  $C_c C_r \cdot e^{-\Delta\alpha l}$  corresponds to C in Formula 3-1 and to suppress  $R_{\max}^{P-P}$  less than 0.1 dB

$$20 \log |C_c \cdot C_r \cdot e^{-\Delta\alpha l}| \leq -44 \text{ dB} \quad (3-2)$$

must be fulfilled.

### 3.2 Ripple Period

Ripple period of this kind of ripple is determined by  $\Delta\beta$  and  $l$ .

Since

$$\frac{d(\Delta\beta x)}{df} = B_x = \frac{2\pi}{C} \left\{ \frac{1}{\sqrt{1-(f_{cx}/f)^2}} - \frac{1}{\sqrt{1-(f_{c01}/f)^2}} \right\} \quad (3-3)$$

$f_{c01}$ ; cut off frequency of  $TE_{01}$  mode

$f_{cx}$ ; cut off frequency of unwanted mode

$C$ ; velocity of light  $(= \frac{1}{\sqrt{\epsilon_0 \mu_0}} = 2.98 \times 10^8 \text{ m/sec})$

ripple period  $f_{px}$  can be shown to be:

$$f_{px} = \frac{2\pi}{B_x \cdot l} \quad (3-4)$$

The values<sup>1</sup> of  $|f_{px} \cdot l| (= \frac{2\pi}{B_x})$  for critical spurious modes are shown in Figure 3-2 and the actual period of  $TE_{02}$  mode and  $e^{-\Delta\alpha l}$  for different lengths are shown in Figure 3-3.

### 3.3 Contribution to Phase Stability of Pilot Signals

When the length of waveguide line changes with the temperature change, mode conversions that are mentioned above affect stability.

Two effects must be considered:

- 1) The phase of 5 MHz and 600 MHz pilot signals transmitted to the antennas through the waveguide are affected by both delay changes and mode-conversion

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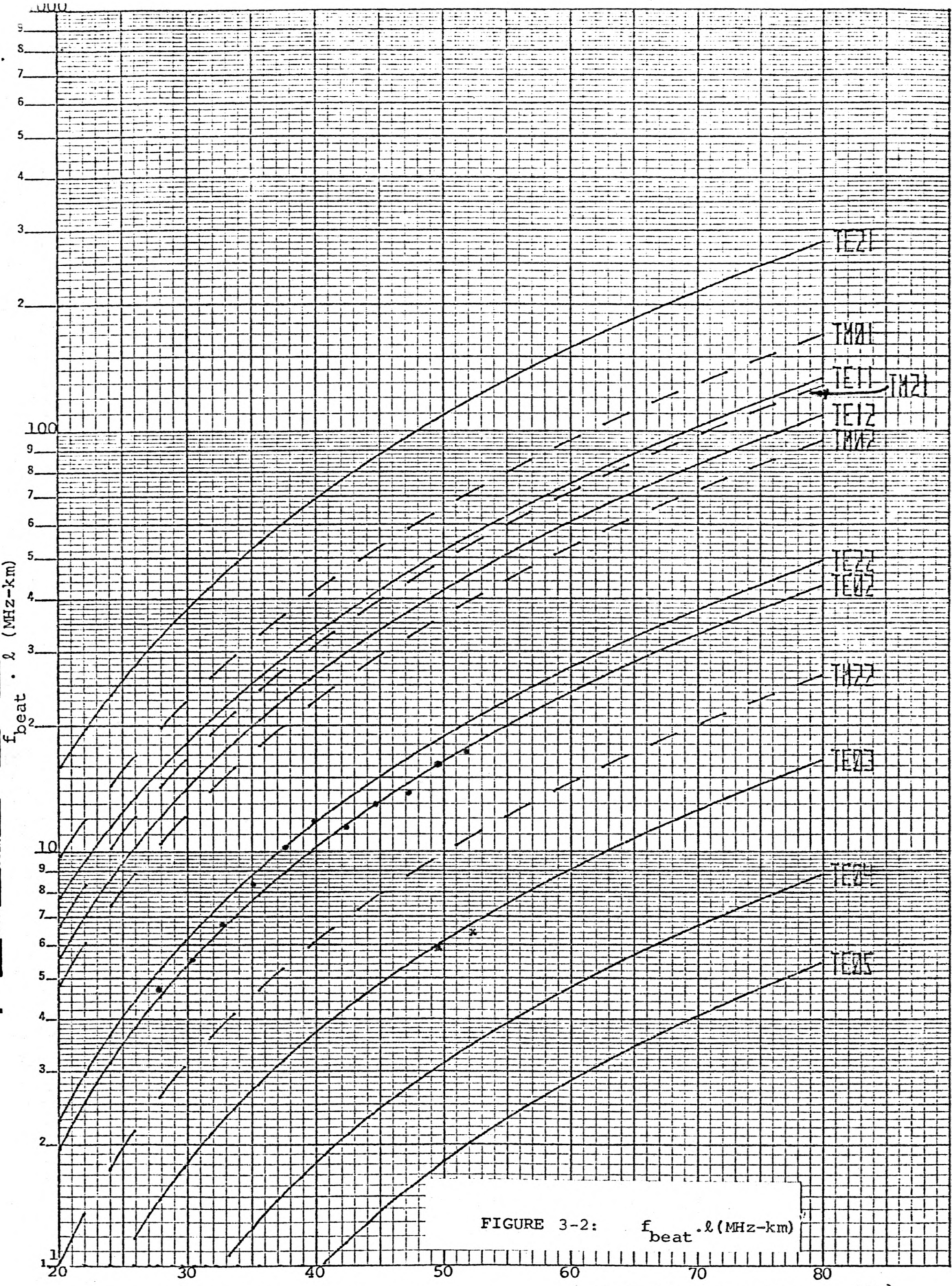


FIGURE 3-2:  $f_{beat} \cdot \lambda$  (MHz-km)

• experimental values of TE<sub>02</sub> mode  
 x experimental values of TE<sub>03</sub> mode

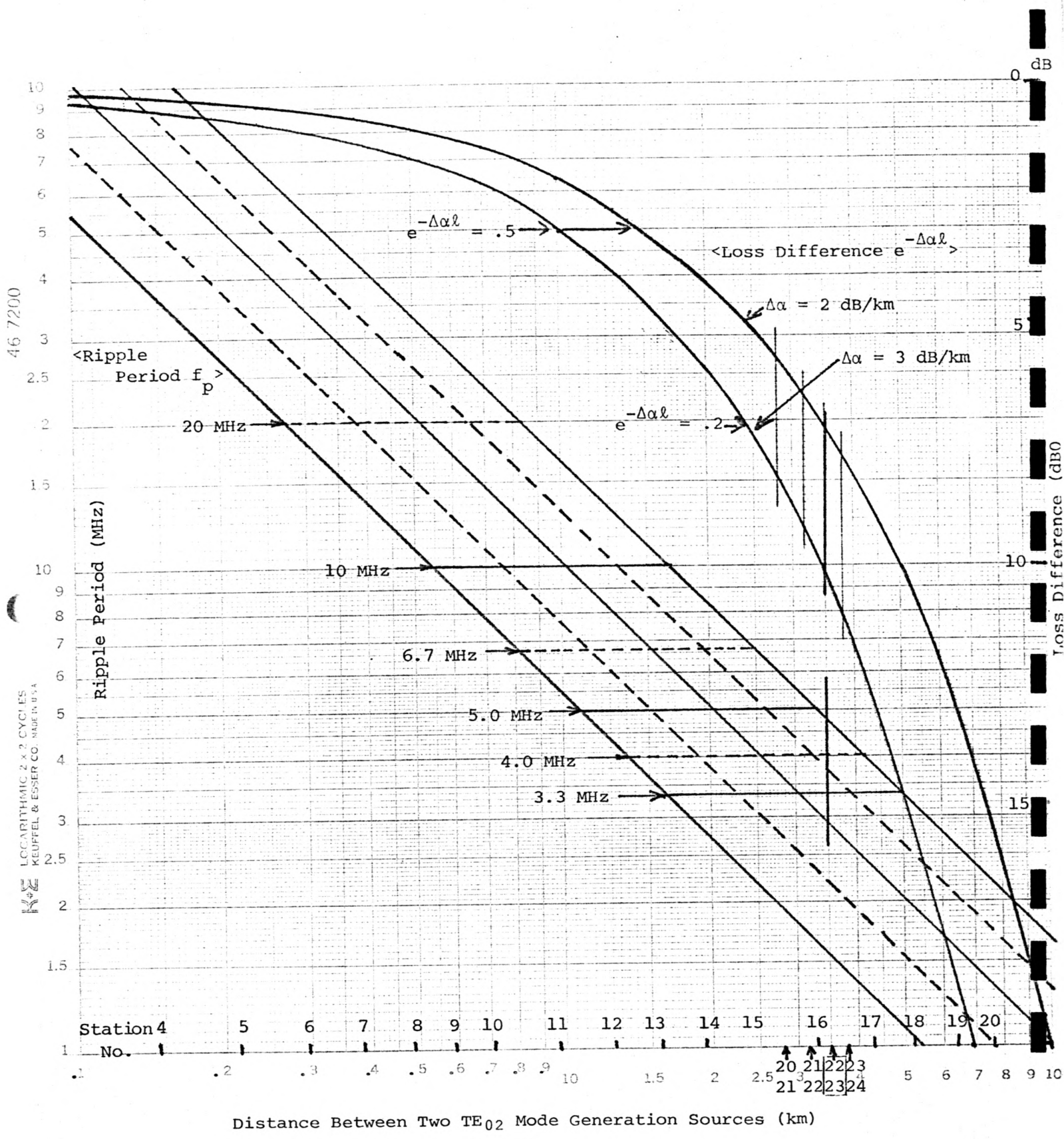


FIGURE 3-3: RIPPLE PERIOD CAUSED BY TE<sub>02</sub> MODE CONVERSION EFFECT



phase changes. However, since the waveguide path is reciprocal, both of the changes are corrected by the L.O. round-trip measuring system. It is worthwhile, however, to calculate how much additional one-way instability is caused by mode conversion.

- 2) The round-trip L.O. correction system will have an offset in the 1 KHz to 100 KHz range between outgoing and return pilot signals. Any phase change which occurs between pilot signals differing by this offset frequency will not be corrected by the round-trip system. Hence it is important to predict the magnitude of this phase change.

Both of the above effects can be analyzed as follows:

$$/A_0 = e^{-\alpha_{01}l} \cdot e^{-j\beta_{01}l} (1 - C_c \cdot C_r \cdot e^{-\Delta\alpha \cdot l} e^{-j\Delta\beta l}) \quad (3-5)$$

$$/A_0' = e^{-\alpha_{01}'l} \cdot e^{-j\beta_{01}'l} (1 - C_c \cdot C_r \cdot e^{-\Delta\alpha' \cdot l} e^{-j\Delta\beta' l}) \quad (3-6)$$

$/A_0, /A_0'$ ; amplitude of  $TE_{01}$  mode shown in vector at frequency  $f_{p1}$  and  $f_{p2}$

$\alpha_{01}, \alpha_{01}'$ ; attenuation constant of  $TE_{01}$  mode at  $f_{p1}$  and  $f_{p2}$

$\beta_{01}, \beta_{01}'$ ; phase constant of  $TE_{01}$  mode at  $f_{p1}$  and  $f_{p2}$

$\Delta\alpha, \Delta\alpha'$ ; the difference in attenuation constant between  $TE_{01}$  mode and spurious mode at  $f_{p1}$  and  $f_{p2}$

$\Delta\beta, \Delta\beta'$ ; the difference in phase constant between  $TE_{01}$  mode and spurious mode at  $f_{p1}$  and  $f_{p2}$

Phase shift caused by mode conversions  $\phi_0$  and  $\phi_0'$  can be shown as follows:

$$\begin{aligned}\phi_0 &= \phi_\ell + \phi_c \\ &= -\beta_{01} \cdot \ell + \sin^{-1}(C_c \cdot C_r \cdot e^{-\Delta\alpha\ell} \cdot \sin\Delta\beta\ell)\end{aligned}\quad (3-7)$$

$$\doteq -\beta_{01} \cdot \ell + C_c \cdot C_r \cdot e^{-\Delta\alpha\ell} \sin\Delta\beta\ell$$

$$\begin{aligned}\phi_0' &= \phi_\ell' + \phi_c' \\ &\doteq -\beta_{01}' \cdot \ell + C_c' \cdot C_r' \cdot e^{-\Delta\alpha'\ell} \sin\Delta\beta'\ell\end{aligned}\quad (3-8)$$

$$(\text{in case of } C_c C_r e^{-\Delta\alpha\ell} \leq 0.01)$$

where

$\phi_0, \phi_0'$ ; phase shift of TE<sub>01</sub> mode at  $f_{p1}$  and  $f_{p2}$

$\phi_\ell, \phi_\ell'$ ; phase shift caused by first order delay

$\phi_c, \phi_c'$ ; phase shift caused by mode conversion

### 3.4 The Instability of Pilot Signals

The phase shift difference  $\Delta\phi_p$  between two pilot frequencies  $f_{p1}$  and  $f_{p2}$  can be calculated as follows:

$$\begin{aligned}\Delta\phi_p &= \phi_0 - \phi_0' \\ &= -(\beta_{01} - \beta_{01}') \cdot \ell + C_c \cdot C_r \cdot e^{-\Delta\alpha \cdot \ell} \sin\Delta\beta\ell - C_c' \cdot C_r' \cdot e^{-\Delta\alpha' \cdot \ell} \sin\Delta\beta'\ell\end{aligned}\quad (3-9)$$

When  $f_{p1}$  and  $f_{p2}$  are close enough to assume  $C_c = C_c'$ ,  $C_r = C_r'$ ,  $\Delta\alpha = \Delta\alpha'$ ,  $\Delta\phi_p$  becomes:

$$\begin{aligned}\Delta\phi_p &= \Delta\phi_\ell + \Delta\phi_c \\ &= -(\beta_{01} - \beta_{01}') \ell + C_c \cdot C_r \cdot e^{-\Delta\alpha\ell} (\sin\Delta\beta\ell - \sin\Delta\beta'\ell) \\ &= -(\beta_{01} - \beta_{01}') \ell + C_c \cdot C_r \cdot e^{-\Delta\alpha\ell} \cdot 2 \cos \frac{\Delta\beta\ell + \Delta\beta'\ell}{2} \sin \frac{\Delta\beta\ell - \Delta\beta'\ell}{2}\end{aligned}\quad (3-10)$$

where

$\Delta\phi_l$ ; phase difference derived from the first order delay between  $f_{p1}$  and  $f_{p2}$

$\Delta\phi_c$ ; phase difference derived from mode conversion effect between  $f_{p1}$  and  $f_{p2}$

Using Formula (3-3) and (3-4),  $\Delta\phi_c$  can be expressed as follows:

$$\Delta\phi_c = C_c \cdot C_r e^{-\Delta\alpha l} \cdot \cos \frac{(\Delta\beta + \Delta\beta')l}{2} \sin \frac{\Delta f_p}{f_{px}} \cdot \pi \quad (3-11)$$

where

$\Delta f_p$ ; frequency difference between two pilot signals

$f_{px}$ ; ripple period around pilot signals ( $\frac{2\pi}{B_x l}$ )

Therefore, the mode conversion that causes the ripple whose period is  $f_{px} = \frac{\Delta f_p}{m}$  ( $m = 1, 2, \dots$ ) does not affect the stability of L.O. measuring system. As our  $f_{px}$  and  $C_c \cdot C_r e^{-\Delta\alpha l}$  can be got from loss measurement,  $\Delta\phi_c$  can be presumed from this formula.

$\Delta\phi_c$  can also be shown in graph as a function of distance  $l$ . One example is shown in Figure 3-4. It shows, even the same amount of the distance change happens, the change of  $\Delta\phi_c$  is different from distance to distance and depends upon the initial distance. The smaller mechanical ripple period  $l_1$

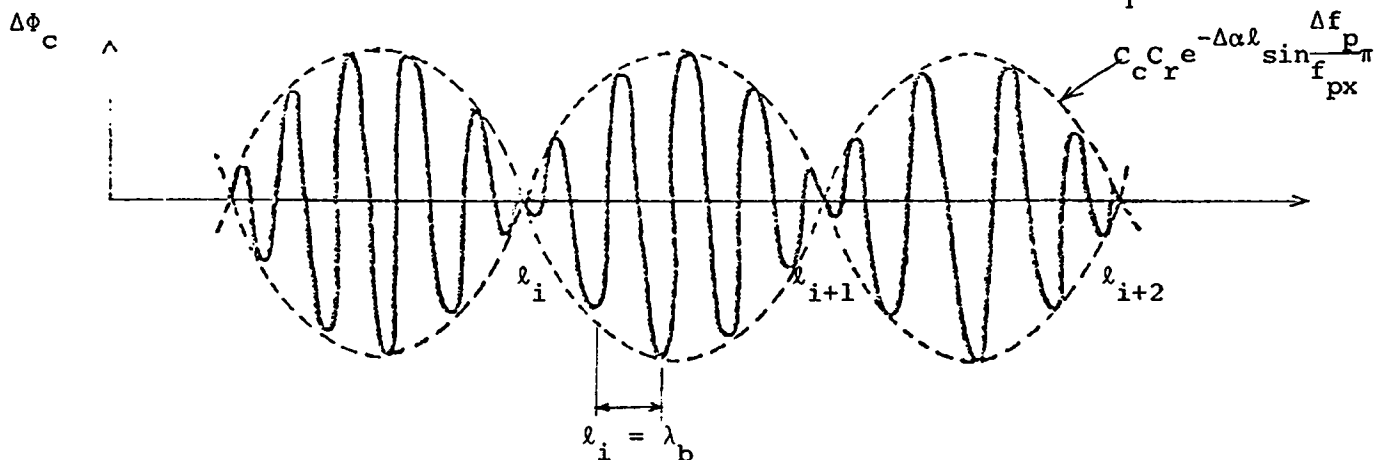


FIGURE 3-4

can be expressed as follows:

$$l_1 = \frac{4\pi}{\Delta\beta + \Delta\beta'} \quad (3-12)$$

$$\doteq \frac{2\pi}{\Delta\beta} = \lambda_b \quad (\text{beat wavelength between TE}_{01} \text{ mode and spurious mode, shown in Figure 3-6})$$

The distances that offer  $\phi_c = 0$  can be expressed as follows:

$$\begin{aligned} l_i &= \frac{2\pi}{\Delta\beta - \Delta\beta'} \quad i \quad (i = 1, 2, \dots) \\ &= \frac{2\pi}{B_x \Delta f_p} \quad i \end{aligned} \quad (3-13)$$

As shown in Figure 3-2,  $\frac{2\pi}{B_x}$  depends upon frequency and spurious mode and it is 7.8 (MHz·Km) for TE<sub>02</sub> mode at 35 GHz in 60 mm waveguide.

When pilot signals of both 5 MHz and 600 MHz are concerned,  $|l_{i+1} - l_i|$  becomes more than 7.5 km and much bigger than  $l_1$ .

Therefore, when one spurious mode generation exists at the station, the channel that has the frequency close to the frequency fulfilling the following formula should be used.

$$\frac{2\pi}{B_x} = \frac{\Delta f_p l_s}{i} \quad (i = 1, 2, \dots) \quad (3-14)$$

where

$l_s$ ; the distance between the station and another spurious mode generation source

In such a case, the distance change from  $l_s$  to  $l_s + \Delta l$  causes the following phase difference change  $\Delta\Delta\phi_c$ :

$$\begin{aligned} \Delta\Delta\phi_c &= 2 \cdot C_c \cdot C_r \cdot e^{-\Delta\alpha l} \cos \frac{(\Delta\beta + \Delta\beta')(l_s + \Delta l)}{2} \cdot \sin \frac{(\Delta\beta - \Delta\beta')(l_s + \Delta l)}{2} \\ &\doteq (-1)^i C_c \cdot C_r \cdot e^{-\Delta\alpha l} \cos \left[ \frac{(\Delta\beta + \Delta\beta')(l_s + \Delta l)}{2} \right] \cdot (\Delta\beta - \Delta\beta') \Delta l \end{aligned} \quad (3-15)$$

$\Delta\Delta\phi_c$  for MHz pilot signals on carrier around 35 GHz becomes as follows:

$$\begin{aligned}\Delta\Delta\phi_c &\leq C_c \cdot C_r e^{-\Delta\alpha\ell} \cdot B_x \Delta f_p \cdot \Delta\ell \\ &\doteq 4.68 \times 10^{-4} \cdot C_c \cdot C_r e^{-\Delta\alpha\ell} \Delta\ell \text{ (deg)}\end{aligned}\tag{3-16}$$

When  $C_c \cdot C_r$  is specified below 0.0057 (-44 dB) this is negligible compared with the contribution by the first order delay shown below:

$$\begin{aligned}\Delta\Delta\phi_\ell &= -(\beta_{01} - \beta_{01}') \Delta\ell \\ &\doteq .012 \Delta\ell \text{ deg } (\Delta\ell; \text{ in mm})\end{aligned}\tag{3-17}$$

But if channel allocation to the station was very bad,  $\Delta\Delta\phi_c$  could increase as follows:

$$\begin{aligned}\Delta\Delta\phi_c &\doteq C_c \cdot C_r e^{-\Delta\alpha\ell} 2\Delta\beta \cdot \Delta\ell \\ &= 4 \cdot C_c \cdot C_r e^{-\Delta\alpha\ell} \cdot \frac{\Delta\ell}{\lambda_b} \pi \\ &\doteq 3.4 \times C_c \cdot C_r \cdot e^{-\Delta\alpha\ell} \cdot \Delta\ell \text{ (deg)}\end{aligned}\tag{3-18}$$

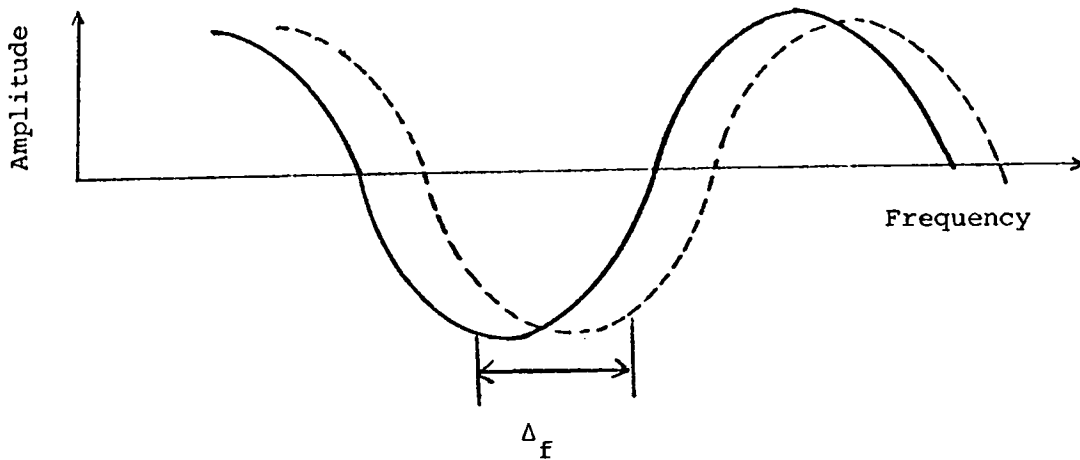
This could be comparable with  $\Delta\Delta\phi_\ell$  in actual case. It means when the distance is carefully set, the effect of first delay could be reduced by mode conversion.

### 3.5 Instability of Amplitude Response

If mode conversion sources exist in line, the distance change between these sources does not only cause affect to the stability of the L.O. system but also to the amplitude response. The amplitude response to frequency involving mode conversion effect has ripples of which properties are shown

in 3-1 and 3-2. It can be shown on Figure 3-5. When the distance becomes longer by  $\Delta l$  the frequency response shifts toward lower frequency as shown in Figure 3-5. And the relationship between  $\Delta f$  and  $\Delta l$  can be shown as follows:

$$\Delta f = \frac{\Delta \beta}{\beta} \cdot \frac{\Delta l}{l} \quad (3-19)$$



And the change  $\Delta l_p$  that gives  $\Delta f = f_{px}$  can be expressed as follows:

$$\Delta l_p = \frac{2\pi}{\Delta \beta} (= \lambda_b) \text{ (beat wavelength between TE}_{01} \text{ and spurious mode, shown in Figure 3-6)} \quad (3-20)$$

Effect of 1-100 KHz offset frequency will be discussed in Appendix A.

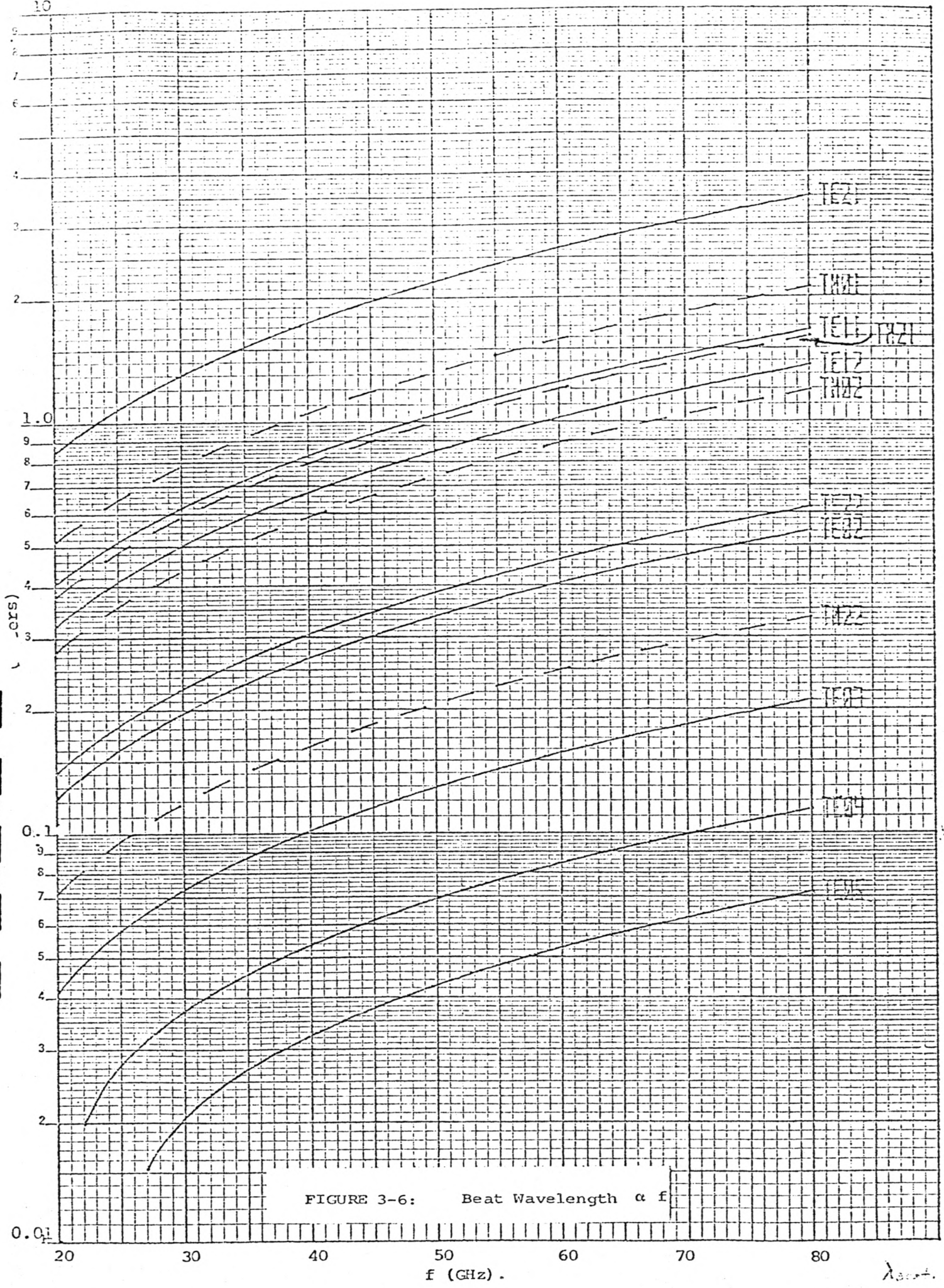


FIGURE 3-6: Beat Wavelength α f

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λ<sub>beat</sub>

#### 4. TE<sub>02</sub> MODE COUPLING EFFECT IN HELICAL COUPLER

A helical coupler couples not only to TE<sub>01</sub> mode but also to TE<sub>on</sub> mode. This effect can be considered similarly to mode conversion effect as explained in Section 3.

##### 4.1 Ripple Amplitude

Output to rectangular waveguide is shown as follows:

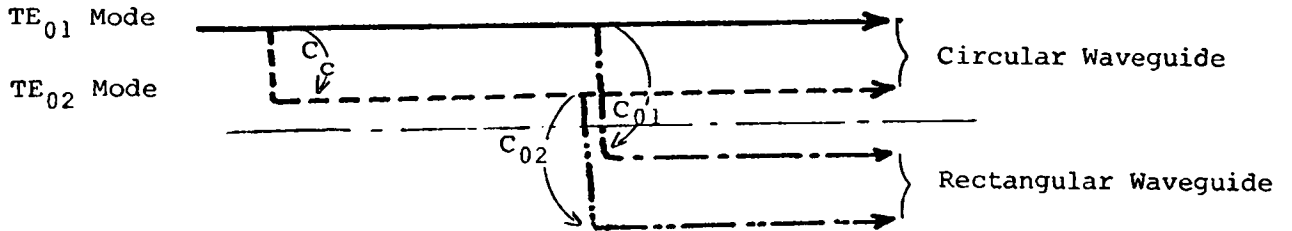


FIGURE 4-1

$$A = e^{-\alpha_{01} \cdot l} \cdot e^{-j\beta_{01} \cdot l} \cdot C_{01} \left\{ 1 - C_c \cdot e^{-\Delta\alpha_{02} l} \left( \frac{C_{02}}{C_{01}} \right) e^{j\Delta\beta_{02}} \right\} \quad (4-1)$$

$C_c$ ; TE<sub>02</sub> mode generation at taper waveguide

$C_{01}, C_{02}$ ; TE<sub>01</sub>, TE<sub>02</sub> mode coupling in helical coupler

$\alpha_{01}$ ; TE<sub>01</sub> mode loss (shown in amplitude)

$\Delta\alpha_{02}$ ; difference in attenuation constant between TE<sub>01</sub> mode and TE<sub>02</sub> mode

$\Delta\beta_{02}$ ; difference in phase constant between TE<sub>01</sub> mode and TE<sub>02</sub> mode

$l$ ; distance between taper waveguide or other TE<sub>02</sub> mode generation source and helical coupler

In this case  $C_c \cdot e^{-\Delta\alpha \cdot l} \cdot \left( \frac{C_{02}}{C_{01}} \right)$  corresponds to C in Formula 1-1 and to suppress  $R_{\max}^{P-P}$  less than 0.1 dB

$$20 \log \left| C_c \cdot e^{-\Delta\alpha \cdot l} \cdot \left( \frac{C_{02}}{C_{01}} \right) \right| \leq -44 \text{ dB} \quad (4-2)$$

is required. In the conventional design of helical coupler



$$20 \log \left| \frac{C_{02}}{C_{01}} \right| \approx -10 \text{ dB}$$

at the edges of 1 GHz frequency band.

In the case of using such helical couplers

$$\left| 20 \log C_c + 20 \log (e^{-\Delta\alpha \cdot l}) \right| \geq 34 \text{ dB}$$

is required.

#### 4.2 Ripple Period

The same argument to that presented in Section 3.2 is valid in this case.

#### 4.3 Contribution to Pilot Signal Phase Stability

The similar arguments to that presented in Section 3.3, 3.4 and 3.5 are valid in this case.

### 5. $TE_{on}$ MODE CONVERSION EFFECT CAUSED BY MANY $TE_{on}$ MODE CONVERSION SOURCES

When there are many spurious mode conversion sources along waveguide lines as shown in Figure 5-1 the attenuation characteristics of  $TE_{01}$  mode can be expressed as follows:<sup>2</sup>

$$A = e^{-\alpha_{01}l - j\beta_{01}l} \cdot e^{-\gamma} \quad (5-1)$$

where

$$\gamma = (1 - a_1 - a_2 \cdots a_n) + \sum_{i,j>i}^n C_i C_j e^{-\Delta\alpha \cdot l_{ij}} \cdot e^{-j\Delta\beta \cdot l_{ij}} \quad (5-2)$$

$a_1, a_2 \cdots a_n$ ; insertion loss of waveguide components  
such as couplers or tapers

$L$  = total length

$l_{ij}$ ; the distance between  $i$  th component and  $j$  th  
component

The second term in Formula 5-2 causes ripple in attenuation,  
and variation of attenuation,  $\sigma^2$ , is given by:

$$\sigma^2 = \frac{1}{2} \sum_{i,j>i}^n \{C_i \cdot C_j e^{-\Delta\alpha l_{ij}}\}^2 \quad (\text{neper}^2) \quad (5-3)$$

And maximum variation of this attenuation  $R_{\max}^{P-P}$  can be  
shown to be:

$$R_{\max}^{P-P} = 2 \sum_{i,j>i}^n |C_i \cdot C_j| \cdot e^{-\Delta\alpha l_{ij}} \quad (\text{neper}) \quad (5-4)$$

When mode conversion amounts of all sources are equal as  
follows:

$$C_1 = C_2 = \cdots = C_n = C_0 \quad (5-5)$$

those values are simplified as follows:

$$\sigma^2 = \frac{1}{2} \sum_{i,j>i}^n (e^{-\Delta\alpha l_{ij}})^2 \cdot C_0^4 \quad (\text{neper}^2) \quad (5-6)$$

$$R_{\max}^{P-P} = 2 \sum_{i,j>i}^n e^{-\Delta\alpha l_{ij}} \cdot C_0^2 \quad (\text{neper}) \quad (5-7)$$

MODE GENERATION					
SOURCE NUMBER $i$	1	2	3	4	5
	AW5	AW6	AW7	AW8	AW9
DISTANCE FROM THE CENTER (Km)	7.66	10.47	13.64	17.16	21.0
	$x_{12}$	$x_{23}$	$x_{34}$	$x_{45}$	
MODE CONVERSION COEFFICIENT BETWEEN $TE_{01}$ AND $TE_{02}$ MODES	$C_1$	$C_2$	$C_3$	$C_4$	
CHANNEL ASSIGNED	$M_4$	$M_3$	$M_2$	$M_1$	$M_0$
COUPLING VALUE	$C_{05}$	$C_{06}$	$C_{07}$	$C_{08}$	0

<Some possibility on design and channel allocation>

1. Channel  $M_0$  and  $M_1$  should be adjacent to each other when it is hard to get sufficient  $TE_{02}$  mode suppression in wide band.
2. To decrease coupled power loss,  $M_0$  and  $M_1$ , should not be adjacent to each other.

FIGURE 5-1: DIAGRAM FOR  $TE_{02}$  MODE CONVERSION EFFECT

When the specification for the total transmission performance is available in the term of  $\sigma^2$  or  $R_{\max}^{P-P}$ , the requirement for  $C$  or maximum allowable  $n$  can be determined.

As an actual case in the VLA waveguide system, couplers that generate some amount of  $TE_{02}$  mode at stations AW5 to AW8 on the southwest arm can be considered. In this case  $\sigma^2$  and  $R_{\max}^{P-P}$  can be written as follows with some assumption of  $\Delta\alpha$ .

$$\sigma^2 = 0.42 \cdot C_0^4 \quad (\text{in case of } \Delta\alpha = 2 \text{ dB/km}) \quad (\text{neper})$$

$$R_{\max}^{P-P} = 17.3 \times 2.04 \times C_0^2 \quad (\text{in case of } \Delta\alpha = 2 \text{ dB/km}) \quad (\text{dB})$$

When the specifications are  $4\sigma \leq 0.1 \text{ dB}$  or  $R_{\max}^{P-P} \leq 0.1 \text{ dB}$ , the maximum allowable  $C_0$  is:

$$C_0 \leq 0.067 \quad (-23.5 \text{ dB}) \quad \text{for } 4\sigma \leq 0.1 \text{ dB}$$

$$C_0 \leq 0.053 \quad (-25.5 \text{ dB}) \quad \text{for } R_{\max}^{P-P} \leq 0.1 \text{ dB}$$

Maximum allowable  $C_0$  for the same requirement in different cases is shown in Figure 5-3.

The above argument is applicable for the evaluation of Michelson interferometer couplers\* that look promising as tight couplers but generate  $TE_{on} (n \geq 2)$  modes.

Judging from Figure 5-3 and some other preliminary test results of this coupler, it looks possible to have at least three of these on one arm. But it may be necessary to suppress  $TE_{02}$  mode to have more than four of these couplers.

\*also referred to as beamsplitter couplers

n	Combination Of TE <sub>02</sub> Mode Generation Sources	The Distance Between Two Sources	Loss Difference In Main Line (e <sup>-Δαl</sup> )		Remarks
			Δα=2 dB/km	Δα=3 dB/km	
1.	AW5-AW6	2.81 km	0.53	0.38	
2.	AW6-AW7	3.17 km	0.49	0.33	
3.	AW7-AW8	3.52 km	0.45	0.30	
4.	AW5-AW7	5.99 km	0.25	0.13	
5.	AW6-AW8	6.68 km	0.21	0.10	
6.	AW5-AW8	9.49 km	0.11	0.03 <sup>8</sup>	
Case I.	Couplers*are installed at AW5-8	$\sum_{n=1}^6 (e^{-\Delta\alpha l_n})^2$	0.842	0.372	
Case II.	Couplers*are installed at AW6-8	$\sum_{n=2,3,5,6} (e^{-\Delta\alpha l_n})^2$	0.499	0.210	
Case III.	Couplers*are installed at AW7-8	$\sum_{n=3} (e^{-\Delta\alpha l_n})^2$	0.202	0.09	

\*couplers generate TE<sub>02</sub> mode in the main line

#### Maximum Allowable Mode Conversion

##### A. Requirement: $4\sigma < 0.1$ dB

	Δα=2 dB/km	Δα=3 dB/km
Case I.	0.047 (-26.5 dB)	0.058 (-24.7 dB)
Case II.	0.054 (-25.8 dB)	0.067 (-23.5 dB)
Case III.	0.068 (-23.4 dB)	0.083 (-21.6 dB)

##### B. Requirement: $R_{\max}^{P-P} < 0.1$ dB

Case I.	0.053 (-25.5 dB)	0.073 (-22.7 dB)
Case II.	0.068 (-23.4 dB)	0.087 (-21.2 dB)
Case III.	0.107 (-19.4 dB)	0.139 (-17.1 dB)

FIGURE 5-2: EVALUATION OF TE<sub>02</sub> MODE CONVERSION EFFECT

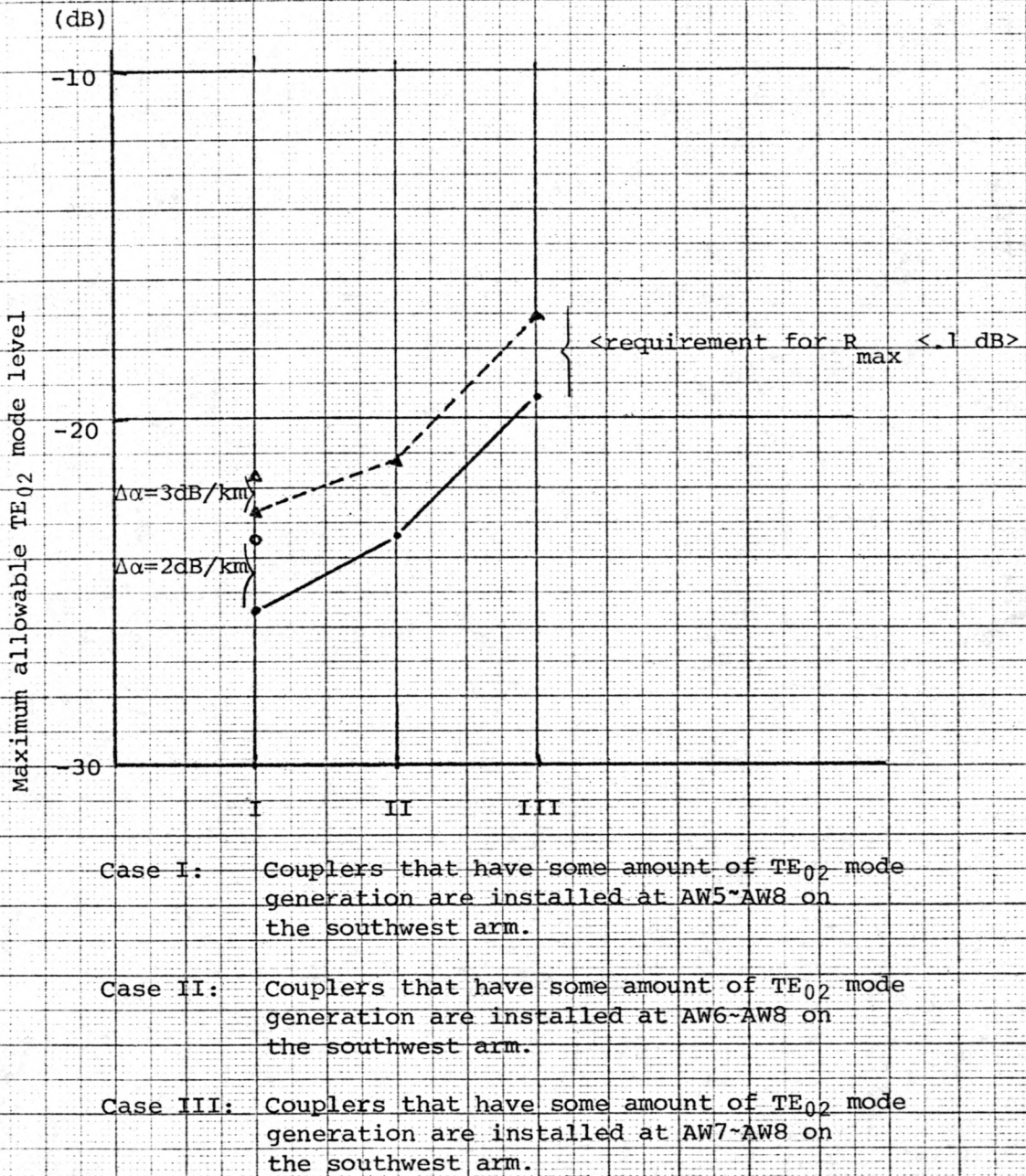


FIGURE 5-3: REQUIREMENT FOR TE<sub>02</sub> MODE SUPPRESSION OF COUPLER INSTALLED AW5-AW8

## 6. LOSS CHANGE AND PILOT SIGNAL PHASE STABILITY UNDER ANTENNA ROTATION

### 6.1 Loss Change

When there are two mode generation sources at both sides of the rotary joint (like the configuration around elevation rotary joint in the case shown in Figure 6-1), the rotation of the joints causes loss changes as shown below even if the rotary joint is perfect:

$$A = e^{-\alpha_0 l} \{1 - C_{f1} \cdot C_{f2} \cdot e^{-\alpha a} \cdot e^{-j\Delta\beta l} \cdot e^{j(n\phi + \phi_1)}\} \quad (6-1)$$

where

$n$ ; the circumferential order of spurious mode  
(unwanted)

$C_{f1}, C_{f2}$ ; mode conversion in flexible waveguide

$20 \log e^{-\alpha a}$ ; spurious mode attenuation between  
two flexible waveguides (dB)  
(is caused by 40" rigid helix waveguide  
in the actual case)

$\alpha_{01}$ ; attenuation constant of  $TE_{01}$  mode

$\Delta\beta$ ; the difference in phase constant between  
spurious mode and  $TE_{01}$  mode

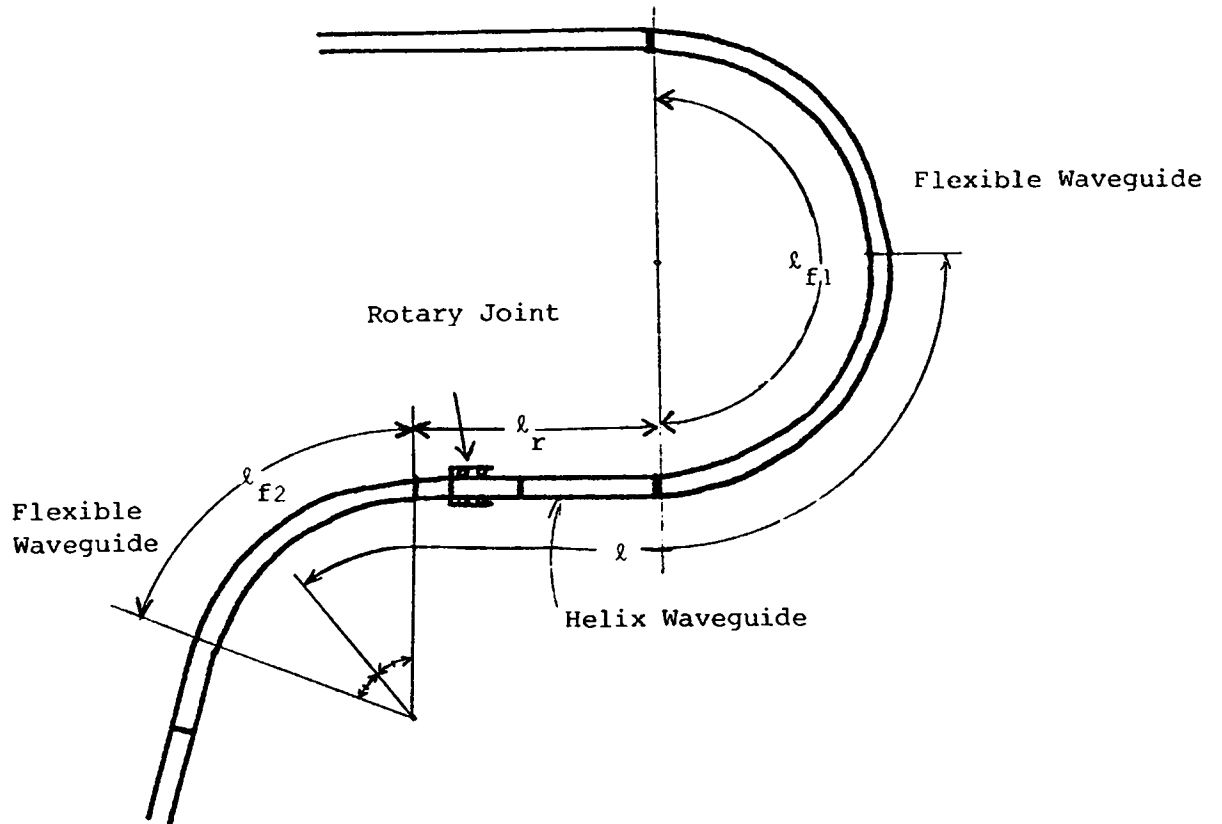
others; shown in Figure 6-1

Usually flexible waveguide generates  $TE_{11}, TE_{12}, TM_{11}$  modes because of its bending and  $TE_{21}, TE_{22}$ , because of its elliptical deformation. These modes are not pure modes in solid waveguide but hybrid modes. Thus  $\Delta\beta$  is different from that in solid waveguide.

In such cases loss variation  $R_{\max}^{P-P}$  becomes as follows:

$$\begin{aligned} R_{\max}^{P-P} &= 20 \log (1 + C_{f1} \cdot C_{f2} \cdot e^{-\alpha a}) - 20 \log (1 - \frac{1}{2} C_{f1} \cdot C_{f2} \cdot e^{-\alpha a}) \\ &\doteq 13.0 C_{f1} \cdot C_{f2} \cdot e^{-\alpha a} \text{ (dB)} \quad (\text{for } n=1) \end{aligned} \quad (6-2)$$

$$\begin{aligned} R_{\max}^{P-P} &= 20 \log (1 + C_{f1} \cdot C_{f2} \cdot e^{-\alpha a}) - 20 \log (1 - C_{f1} \cdot C_{f2} \cdot e^{-\alpha a}) \\ &\doteq 17.3 C_{f1} \cdot C_{f2} \cdot e^{-\alpha a} \text{ (dB)} \quad (\text{for } n=2) \end{aligned} \quad (6-3)$$



$$l = (l_{f1} + l_{f2})/2 + l_r$$

- $\phi_0$ ; angle deviation from initial angle  
 ( $-30^0 \sim 80^0$  in our case)  
 ( $-30^0 \sim 90^0$  is used for analysis)
- $\phi_i$ ; initial angle ( $=180^0$  in above case)
- $l_{f1}, l_{f2}$ ; the length of bend part of flexible waveguides
- $l$ ; equivalent length between two bended waveguides

FIGURE 6-1: ANTENNA WAVEGUIDE CONFIGURATION AROUND ELEVATION ROTARY JOINT



When the bend or elliptical deformation is uniform along the bent sections, mode conversion amount  $C_{f1}$ ,  $C_{f2}$  depends upon frequency and length of flexible waveguides and is given as follows:<sup>3</sup>

$$C_{f1} = C_0 \sin(\Delta\beta l_{f1}/2) \quad (6-3)$$

$$C_{f2} = C_0 \sin(\Delta\beta l_{f2}/2) \quad (6-4)$$

where

$C_0$ ; maximum conversion

This means that at certain frequencies  $C_{f1}$  and  $C_{f2}$  become zero and loss variation does not occur even under rotation of antenna. The frequency period that has no loss change is defined as  $f_p/2$  for later.

## 6.2 Pilot Signal Phase Stability

Mode conversions at both sides of the rotary joint cause not only loss change but also phase difference change of the two pilot signals. The latter can be analyzed as follows:

$$\phi_r = \sin^{-1}\{C_c \cdot C_r \cdot \sin\Delta\beta l \cdot \sin(n\phi_0 + \phi_i)\} \quad (6-5)$$

$$\doteq C_c \cdot C_r \cdot \sin\Delta\beta l \cdot \sin(n\phi_0 + \phi_i)$$

$$\phi_r' \doteq C_c' \cdot C_r' \cdot \sin\Delta\beta' l \cdot \sin(n\phi_0 + \phi_i) \quad (6-6)$$

$$\begin{aligned} \Delta\phi_r &= \phi_r - \phi_r' \\ &= \sin(n\phi_0 + \phi_i) \{C_c \cdot C_r \cdot \sin\Delta\beta l - C_c' \cdot C_r' \cdot \sin\Delta\beta' l\} \quad (6-7) \end{aligned}$$

where

$\phi_r, \phi_r'$ ;  $TE_{01}$  mode phase shift caused by mode conversions in both sides of rotary joint at frequency  $f_{p1}$  and  $f_{p2}$  on which two pilot signals are sent.

In the case of 5 MHz pilot signals,  $C_c = C_c'$  and  $C_r = C_r'$  can be assumed and  $\Delta\phi_r$  becomes as follows:

$$\begin{aligned}\Delta\phi_r &= \text{Sin}(n\phi_0 + \phi_i) \cdot C_c \cdot C_r \{ \text{Sin}\Delta\beta l - \text{Sin}\Delta\beta' l \} \\ &\doteq \text{Sin}(n\phi_0 + \phi_i) \cdot C_c \cdot C_r \cdot (\Delta\beta - \Delta\beta') \cdot l\end{aligned}\quad (6-8)$$

Maximum deviation  $\Delta\Delta\phi_{r\text{max}}^{\text{P-P}}$  becomes as follows in the area of  $-30^\circ \leq \phi_0 \leq 90^\circ$ ; ( $\phi_i = +180^\circ$ )

$$\Delta\Delta\phi_{r\text{max}} = 1.5 \cdot C_c \cdot C_r \cdot (\Delta\beta - \Delta\beta') \cdot l \quad (\text{for } n=1) \quad (6-9)$$

$$\Delta\Delta\phi_{r\text{max}} = 2 \cdot C_c \cdot C_r \cdot (\Delta\beta - \Delta\beta') \cdot l \quad (\text{for } n=2) \quad (6-10)$$

From Formulas (6-2) and (6-3),

$C_c \cdot C_r$  can be derived from  $R_{\text{max}}^{\text{P-P}}$  as follows:

$$C_c \cdot C_r = \frac{R_{\text{max}}^{\text{P-P}}}{13.0} \quad (\text{for } n=1) \quad (6-11)$$

$$C_c \cdot C_r = \frac{R_{\text{max}}^{\text{P-P}}}{17.3} \quad (\text{for } n=2) \quad (6-12)$$

From the definition of  $f_p$

$(\Delta\beta - \Delta\beta') \cdot l$  can be shown to be:

$$(\Delta\beta - \Delta\beta') l = \frac{\Delta f_p}{f_p} \cdot 2\pi \quad (\text{rad}) \quad (6-13)$$

where

$f_p$ ; the frequency period of no loss variation under antenna rotation (150-2000 MHz)

$\Delta f_p$ ; the frequency difference between two pilot signals

From (6-9) and (6-10)

Thus  $\Delta\Delta\phi_{r\text{max}}$  can be written as follows:

$$\Delta\Delta\phi_{r\text{max}} = 3.1 \times 10^2 \frac{R_{\text{max}}^{\text{P-P}}}{f_p} \quad (\text{degree}) \quad (6-14)$$

The measured values of  $R_{\max}^{P-P}$  and  $f_p$  on No. 2 antenna are 0.1-0.2 dB<sup>P-P</sup> and 150-2000 MHz. So in the worst case

$$\Delta\Delta\phi_{r\max\ worst} = 0.41 \quad (\text{degrees}) \quad (6-15)$$

In the case of 600 MHz pilot signals

$$\begin{aligned} \Delta\Delta\phi_r &\leq 1.5 \cdot (C_c \cdot C_r + C_c' \cdot C_r') \\ &= 3.12 \cdot (R_{\max}^{P-P} + R_{\max}^{P-P'}) \quad (\text{deg}) \quad (6-16) \\ R_{\max}^{P-P}, R_{\max}^{P-P'} &; \text{maximum variation caused by antenna} \\ &\text{rotation at } f_p \text{ and } f_p' \end{aligned}$$

To maintain  $\Delta\Delta\phi_r$  less than 1 degree in the worst case, the following is required:

$$R_{\max}^{P-P} + R_{\max}^{P-P'} \leq 0.32 \quad (\text{dB}) \quad (6-17)$$

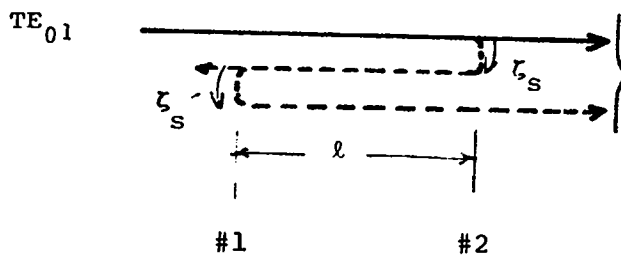
## 7. $TE_{01}$ MODE REFLECTION EFFECT

### 7.1 Reflection Sources

Reflection sources in  $TE_{01}$  mode circular waveguide have many different properties from rectangular waveguides or  $TE_{11}$  mode circular waveguide. In the usual  $TE_{01}$  mode waveguides, dimensional imperfections cause forward-going unwanted mode waves, but very few  $TE_{01}$  mode reflections. The main sources of  $TE_{01}$  mode reflection are circular-rectangular adapters and couplers that are inserted in the main line.

### 7.2 Ripple Amplitude

When there are two reflection sources as shown in Figure 7-1, the output  $TE_{01}$  mode can be given as follows:



#1, #2, reflection sources

FIGURE 7-1

$$A = e^{-\alpha_{01}\ell} (\sqrt{1-\zeta_s^2} \cdot \sqrt{1-\zeta_s'^2} + \zeta_s \zeta_s' e^{-2\alpha_{01}\ell} \cdot e^{-2j\beta_{01}\ell}) \quad (7-1)$$

$$\doteq e^{-\alpha_{01}\ell} (1 + \zeta_s \zeta_s' e^{-2\alpha_{01}\ell} \cdot e^{-2j\beta_{01}\ell})$$

where

$\zeta_s, \zeta_s'$ ; reflection co-efficient

$\alpha_{01}$ ; attenuation constant of  $TE_{01}$  mode

$\beta_{01}$ ; phase constant of  $TE_{01}$  mode

$\ell$ ; the distance between two reflection sources

In such cases,  $\zeta_s \cdot \zeta_s' \cdot e^{-2\alpha_{01}\ell}$  corresponds to spurious signal level C shown in Formula (1-1).

To suppress peak-peak variation of attenuation less than 0.1 dB, the following requirement must be filled:

$$20 \log | \zeta_s \cdot \zeta_s' \cdot e^{-2\alpha_{01}\ell} | \leq -44 \text{ dB} \quad (7-2)$$

### 7.3 Ripple Period

Ripple period of this kind of ripple is determined by  $TE_{01}$  mode phase constant  $\beta_{01}$  and the distance  $\ell$  between reflection sources, and can be expressed as follows:

$$f_p = \frac{C}{2\ell} = \frac{2\pi}{2 \cdot \ell \frac{d\beta_{01}}{df}} \quad (7-3)$$

The figures for C at different frequencies are shown in Table 7-1. The higher the operational frequency becomes from cut-off frequency, the closer constant C becomes to 300 (MHz-m).

An antenna waveguide is about 40 m long and when there are reflections at both ends of the antenna waveguide the ripple period is approximately 3.75 MHz.

When this amplitude is too large, it might cause some problem in observation of spectrum line radio sources. This means that specification of  $TE_{01}$  mode reflection in rectangular-

Frequency (GHz)	CONSTANT C (MHz m)	
	60 mm Waveguide	20 mm Waveguide
30	304.4	376.0
35	302.8	349.6
40	301.6	335.1
45	300.7	326.2
50	300.2	320.2

TABLE 7-1: MULTIPLE OF RIPPLE PERIOD AND  
DISTANCE BETWEEN TWO REFLECTION SOURCES

circular adapter, modem input is very important. These should be suppressed to under - 25 dB.

In main line, as the distance between reflection sources becomes much longer except at the center of the "Y", the ripple period becomes much smaller.

#### 7.4 Contribution to Phase Stability of Pilot Signals

When the length of waveguide line changes with the temperature, the ripple that arises from TE<sub>01</sub> mode reflection affects the phase stability of pilot signals (together with other factors such as the first delay change or mode conversion effects). This contribution can be expressed as follows:

$$A_0 = e^{-\alpha_{01} \cdot l} \cdot e^{-j\beta_{01}l} (1 - \zeta_{st} \zeta_{sr} e^{-2\alpha_{01}l} \cdot e^{-2j\beta_{01}l}) \quad (7-4)$$

$$A_0' = e^{-\alpha_{01}' \cdot l} \cdot e^{-j\beta_{01}'l} (1 - \zeta_{st}' \zeta_{sr}' e^{-2\alpha_{01}'l} \cdot e^{-2j\beta_{01}'l}) \quad (7-5)$$

where

$A_0, A_0'$ ; amplitude of TE<sub>01</sub> mode shown in vector at frequency  $f_{p1}, f_{p2}$

$\alpha_{01}, \alpha_{01}'$ ; attenuation constant of TE<sub>01</sub> mode at  $f_{p1}, f_{p2}$

$\beta_{01}, \beta_{01}'$ ; phase constant of TE<sub>01</sub> mode at  $f_{p1}, f_{p2}$

$\zeta_{st}, \zeta_{st}'$ ; reflection co-efficient in circular port of transmitting equipment at  $f_{p1}, f_{p2}$

$\zeta_{sr}, \zeta_{sr}'$ ; reflection co-efficient in circular port of receiving equipment at  $f_{p1}, f_{p2}$

Phase shifts  $\phi_0$  and  $\phi_0'$  between transmitting and receiving ports can be shown as follows:

$$\begin{aligned}\phi_0 &= \phi_\ell + \phi_\zeta \\ &= -\beta_{01} \cdot \ell + \sin^{-1}(\zeta_{st} \zeta_{sr} e^{-2\alpha_{01} \cdot \ell} \sin 2\beta_{01} \ell) \quad (7-6) \\ &= -\beta_{01} \cdot \ell + \zeta_{st} \zeta_{sr} e^{-2\alpha_{01} \cdot \ell} \sin 2\beta_{01} \ell\end{aligned}$$

$$\begin{aligned}\phi_0' &= \phi_\ell' + \phi_\zeta' \\ &= -\beta_{01}' \cdot \ell + \zeta_{st}' \cdot \zeta_{sr}' e^{-2\alpha_{01}' \cdot \ell} \sin 2\beta_{01}' \cdot \ell \quad (7-7)\end{aligned}$$

The difference in phase shifts  $\phi_0$ ,  $\phi_0'$  between two pilot frequencies  $f_{p1}$  and  $f_{p2}$  can be expressed as follows:

$$\begin{aligned}\Delta\phi_p &= \phi_0 - \phi_0' \\ &= -(\beta_{01} - \beta_{01}') \cdot \ell + \zeta_{st} \zeta_{sr} e^{-2\alpha_{01} \ell} \sin 2\beta_{01} \ell \quad (7-8) \\ &\quad - \zeta_{st}' \zeta_{sr}' e^{-2\alpha_{01}' \ell} \sin 2\beta_{01}' \ell\end{aligned}$$

When  $f_{p1}$  and  $f_{p2}$  are close enough to assume  $\zeta_{st} = \zeta_{st}'$ ,  $\zeta_{sr} = \zeta_{sr}'$ ,  $\alpha_{01} = \alpha_{01}'$ ,  $\Delta\phi_p$  becomes as follows:

$$\begin{aligned}\Delta\phi_p &= \Delta\phi_\ell + \Delta\phi_\zeta \\ &= -(\beta_{01} - \beta_{01}') \cdot \ell + \zeta_{st} \cdot \zeta_{sr} e^{-2\alpha_{01} \ell} (\sin 2\beta_{01} \ell - \sin 2\beta_{01}' \ell) \quad (7-9) \\ &= -(\beta_{01} - \beta_{01}') \cdot \ell + \zeta_{st} \cdot \zeta_{sr} e^{-2\alpha_{01} \ell} \cdot 2 \cos(\beta_{01} + \beta_{01}') \ell\end{aligned}$$

where  $\sin(\beta_{01} - \beta_{01}') \ell$

$\Delta\phi_\ell$ ; phase difference derived from the first order delay between  $f_{p1}$  and  $f_{p2}$

$\Delta\phi_\zeta$ ; phase difference derived from TE<sub>01</sub> mode reflection effect

Using Formula 7-3  $\Delta\phi_{\zeta}$  can be expressed as follows:

$$\Delta\phi_{\zeta} = \zeta_{st} \cdot \zeta_{sr} \cdot e^{-2\alpha_{01}\ell} \cdot 2 \cos(\beta_{01} + \beta'_{01}) \cdot \ell \sin\left(\frac{\Delta f}{f_p} \cdot \pi\right) \quad (7-10)$$

$\Delta f_p$ ; frequency difference between two pilot signals  
 $f_p$ ; ripple period around pilot signals ( $=\frac{C}{x \cdot \ell}$ )

It shows that the  $TE_{01}$  mode reflections that cause the ripple whose period is  $f_p = \frac{p}{m}$  ( $m=1,2,\dots$ ) do not affect the L.O. system. As  $\zeta_{st} \zeta_{sr} e^{-2\alpha_{01}\ell}$  and  $f_p$  can be known from loss measurement variation of  $\Delta\phi_{\zeta}$  can be presumed from this formula.

$\Delta\phi_{\zeta}$  can also be shown in graph as a function of distance  $\ell$ . It is shown in Figure 7-2.

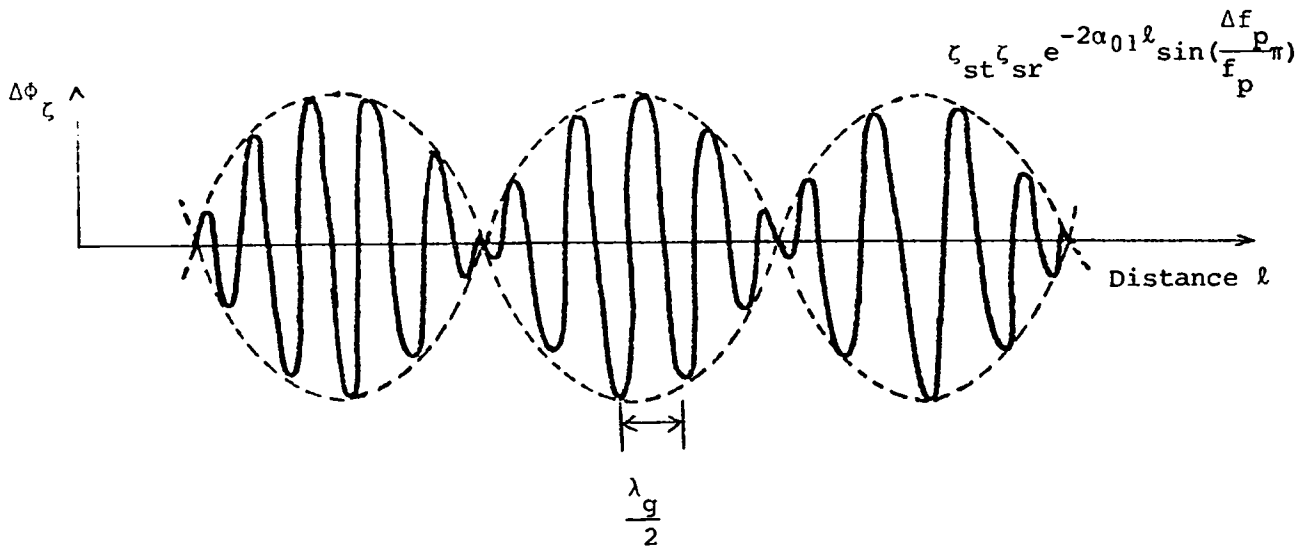


FIGURE 7-2

It shows, even if the same amount of the distance change happens, the change of  $\Delta\phi_{\zeta}$  varies with the initial distance. In the case of 5 MHz pilot signal,  $\ell_2$  is almost 15 m and when the distance between two reflection sources becomes



integer times of this distance, variation of  $\Delta\phi_\zeta$  with distance change  $\Delta\ell$  can be minimized and shown as follows:

$$\begin{aligned} \Delta\Delta\phi_\zeta &= 2 \cdot \zeta_{st} \zeta_{sr} e^{-2\alpha_{01}\ell} \cos(\beta_{01} + \beta_{01}') (\ell_i + \Delta\ell) \\ &\quad \sin(\beta_{01} - \beta_{01}') (\ell_i + \Delta\ell) \quad (7-11) \\ &\doteq (-1)^i \cdot 2 \cdot \zeta_{st} \zeta_{sr} e^{-2\alpha_{01}\ell} \cdot (\beta_{01} - \beta_{01}') \Delta\ell \cdot \cos(\beta_{01} + \beta_{01}') \ell \end{aligned}$$

$\Delta\Delta\phi_\zeta$  for 5 MHz pilot signals can be got as follows:

$$\Delta\Delta\phi_\zeta \leq 2.4 \times 10^{-3} \cdot \zeta_{st} \cdot \zeta_{sr} e^{-\alpha_{01}\ell} \Delta\ell \quad (7-12)$$

But when reflection sources exist with unfavorable distance  $\Delta\Delta\phi_\zeta$  could increase as follows:

$$\Delta\Delta\phi_\zeta \leq 4 \cdot \zeta_{st} \zeta_{sr} e^{-2\alpha_{01}\ell} \quad (7-13)$$

The distance from a coupler to a circular-rectangular adapter in the vertex room is almost 40 m. This is not so bad a length but it seems nice to get pad about 13 m away from the coupler to minimize the unstability in wide frequency range. The pad will also be helpful to reduce the ripple caused between coupler and adapter. The position almost corresponds to the top of the ground.

#### 7.4 Instability of Amplitude Response

Such reflections do not only affect the stability of L.O. system but also affect the stability of amplitude response when the distance between reflection sources changes. The ripple period of which property are explained in 7-1 and 7-2 can be shown in Figure 7-3. When the distance becomes longer by  $\Delta\ell$  the frequency response shifts toward lower frequency as shown in Figure 7-3. The relationship between  $\Delta f$  and  $\Delta\ell$  can be expressed

as follows:

$$\begin{aligned} \Delta f &= \frac{\beta}{\frac{d\beta}{df}} \cdot \frac{\Delta l}{l} \\ &= \frac{C}{2\pi} \cdot \frac{\Delta l}{l} \end{aligned} \quad (7-14)$$

And, the change  $\Delta l_p$  that gives  $\Delta f = f_p$  can be expressed as follows:

$$\Delta l_p = \frac{2\pi}{2\beta} = \frac{\lambda_g}{2} \quad (7-15)$$

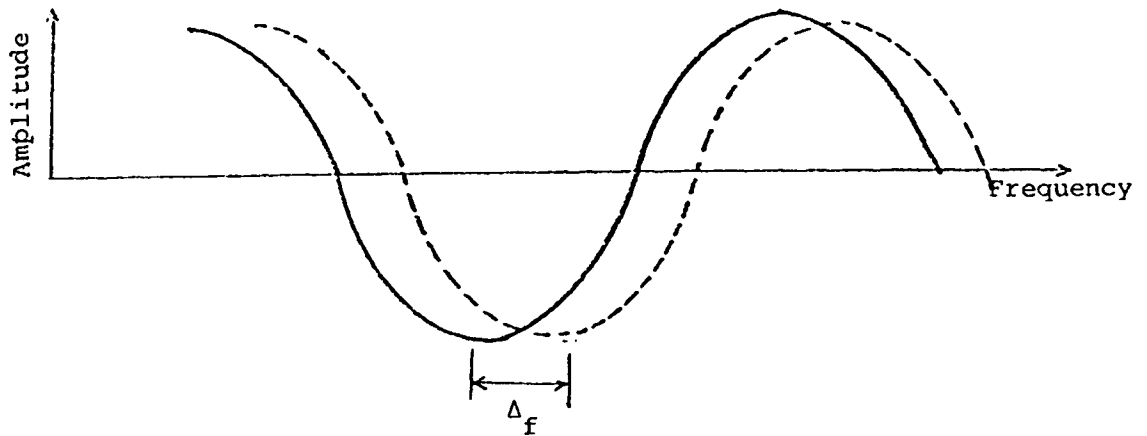


FIGURE 7-3

#### 8. EFFECT OF DIRECTIVITY OF COUPLERS

When some power goes to the reverse direction from main power and it hits reflection sources, it comes back again in the forward direction and becomes a spurious signal to main power. This mechanism is shown in Figure 8-1. In this case, similar effect to

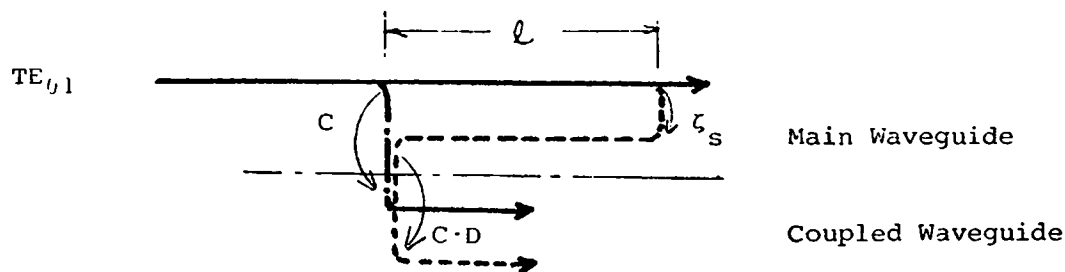


FIGURE 8-1

the reflection effect occurs. The ripple amplitude, period and contribution can be obtained by converting some symbols in the foregoing section as follows:

$$\zeta_s' \rightarrow D; \text{ directivity of coupler}$$

$$\zeta_s \rightarrow \zeta_s$$

This shows the importance of coupler and to suppress the amplitude of this kind of ripples less than 0.1 dB

$$20 \log | \zeta_s \cdot e^{-2\alpha_0 l} \cdot D | \leq -44 \text{ dB}$$

### 9. TE<sub>01</sub> MODE REFLECTION EFFECT CAUSED BY COUPLERS

When there are many TE<sub>01</sub> mode reflection sources such as couplers along waveguide line, shown in Figure 9-1, the attenuation characteristics of  $i$ th coupler can be expressed as follows (by central limit theorem).

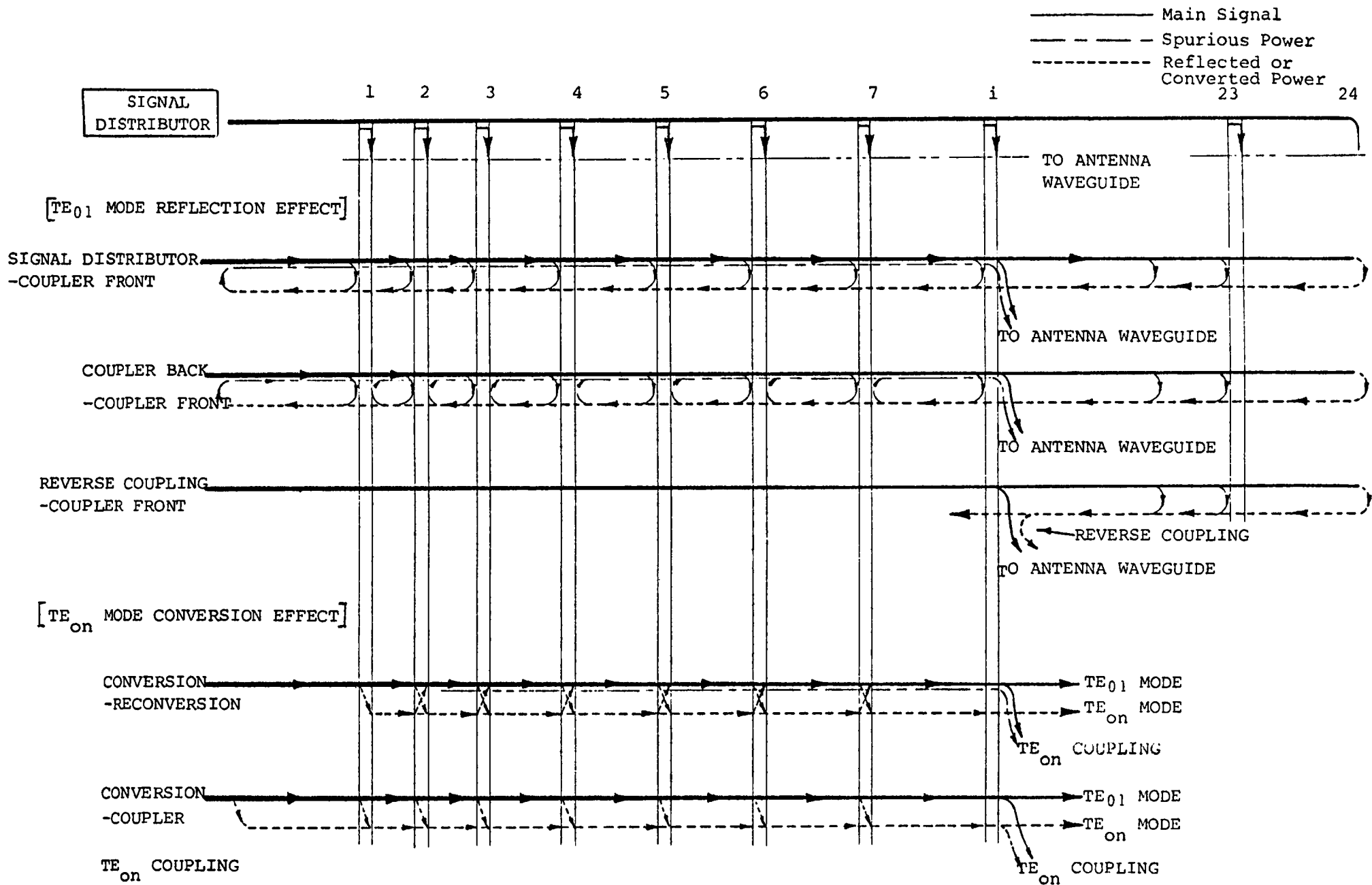


FIGURE 9-1: MULTI-REFLECTION AND MODE CONVERSION EFFECT

$$A = e^{-(\alpha+j\beta)(\ell_0+\ell_i)} e^{-(r_0+r_1+r_2+r_3)} \cdot C_i$$

where

$$r_0 = (1 - a_1 - a_2 - \dots - a_i)$$

$$r_1 = \sum_{j=1}^N \{ \zeta_s \zeta_{fj} e^{-2\alpha(\ell_0+\ell_j)-2\alpha\ell_j} \cdot e^{-2j\beta(\ell_0+\ell_j)} \}$$

$$r_2 = \sum_{j=1}^{i-1} \sum_{k=j+1}^N \{ \zeta_{bj} \zeta_{fk} e^{-2\alpha(\ell_k-\ell_j)-2\alpha\ell_k} e^{-2j\beta(\ell_k-\ell_j)} \}$$

$$r_3 = \sum_{j=i+1}^N \{ D_i \zeta_{fj} e^{-2\alpha(\ell_j-\ell_i)-2\alpha\ell_j} e^{-2j\beta(\ell_j-\ell_i)} \}$$

$a_i$ ; insertion loss of  $i$  th coupler

$\alpha_{jk}$ ; the sum of insertion loss of couplers between  $j$  th and  $k$  th couplers

$$(\alpha_{jk} = \sum_{m=j+1}^{k-1} a_m)$$

$\zeta_s$ ; reflection coefficient in circular port of signal distributor

$\zeta_{fj}$ ; reflection coefficient in front main circular port of  $j$  th coupler

$\zeta_{bj}$ ; reflection coefficient in back main circular port of  $j$  th coupler

$D_i$ ; directivity of  $i$  th coupler

$C_i$ ; coupling of  $i$  th coupler

$\ell_0$ ; the distance between control building and the center of the "Y"

$\ell_i$ ; the distance between the center of the "Y" and  $i$  th antenna station

$\alpha$ ; attenuation constant of  $TE_{01}$  mode

$\beta$ ; phase constant of  $TE_{01}$  mode

The terms  $r_1, r_2, r_3$  causes ripple in attenuation and variation of attenuation  $\sigma^2$  can be shown to be as follows:

$$\sigma^2 = \sigma_1^2 + \sigma_2^2 + \sigma_3^2$$

$$\sigma_1^2 = \frac{1}{2} \sum_{j=1}^S \{\zeta_s \zeta_{fj} e^{-2\alpha(\ell_0 + \ell_j) - 2\alpha_0 j}\}^2$$

$$\sigma_2^2 = \frac{1}{2} \sum_{j=1}^{i-1} \sum_{k=j+1}^S \{\zeta_{bj} \zeta_{fk} e^{-2\alpha(\ell_k - \ell_j) - 2\alpha_k j}\}^2$$

$$\sigma_3^2 = \frac{1}{2} \sum_{j=i+1}^S \{D_i \zeta_{fj} e^{-2\alpha(\ell_j - \ell_i) - 2\alpha_{ij}}\}^2$$

When the couplers have the identical performance as follows:

$$\zeta_{b1} = \zeta_{b2} = \dots = \zeta_{bi} = \dots = \zeta_{b0}$$

$$\zeta_{f1} = \zeta_{f2} = \dots = \zeta_{fi} = \dots = \zeta_{f0}$$

$$D_1 = D_2 = \dots = D_i = \dots = D_0$$

$\sigma_1^2, \sigma_2^2, \sigma_3^2$  can be simplified to be as follows:

$$\sigma_1^2 = \frac{1}{2} (\zeta_s \zeta_{f0})^2 \sum_{j=1}^S \{e^{-2\alpha(\ell_0 + \ell_j) - 2\alpha_0 j}\}^2 = \frac{1}{2} (\zeta_s \zeta_{f0})^2 R_1^2$$

$$\sigma_2^2 = \frac{1}{2} (\zeta_{b0} \zeta_{f0})^2 \sum_{j=1}^{i-1} \sum_{k=j+1}^S \{e^{-2\alpha(\ell_k - \ell_j) - 2\alpha_k j}\}^2 = \frac{1}{2} (\zeta_{b0} \zeta_{f0})^2 R_2^2$$

$$\sigma_3^2 = \frac{1}{2} (D_0 \zeta_{f0})^2 \sum_{j=i+1}^S \{e^{-2\alpha(\ell_j - \ell_i) - 2\alpha_{ij}}\}^2 = \frac{1}{2} (D_0 \zeta_{f0})^2 R_3^2$$

Based on the following assumptions:

$$a_1 = a_{20} = 0.0114 (=0.1 \text{ dB})$$

$$a_{21} = 0.025 \quad (13 \text{ dB coupler})$$

$$a_{22} = 0.051 \quad (10 \text{ dB coupler})$$

$$a_{23} = 0.135 \quad (6 \text{ dB coupler})$$

$$a_{24} = 0$$

$$l_0 = 450 \text{ m}$$

$l_i$ ; corresponds to antenna locations on the West Arm

$$\alpha_{01}; 1.0 \sim 3.0 \text{ dB/km}$$

$$i; 1 \sim 24$$

$R_1^2$ ,  $R_2^2$ ,  $R_3^2$  were calculated for every antenna station of 1st to 24th on the West Arm and tabulated in Figures 9-2 and 9-3.

$R_1^2$  is constant for every antenna station.  $R_2^2$  increases when the antenna station becomes far away from the center, but  $R_3^2$  is smaller near the center.

Requirement for getting  $4\sigma_1 \leq 0.1 \text{ dB}$  under  $\alpha_{01} = 1.2 \text{ dB/km}$  is:

$$\zeta_s \cdot \zeta_{f0} \leq 1.74 \times 10^{-3} \quad (-55.2 \text{ dB})$$

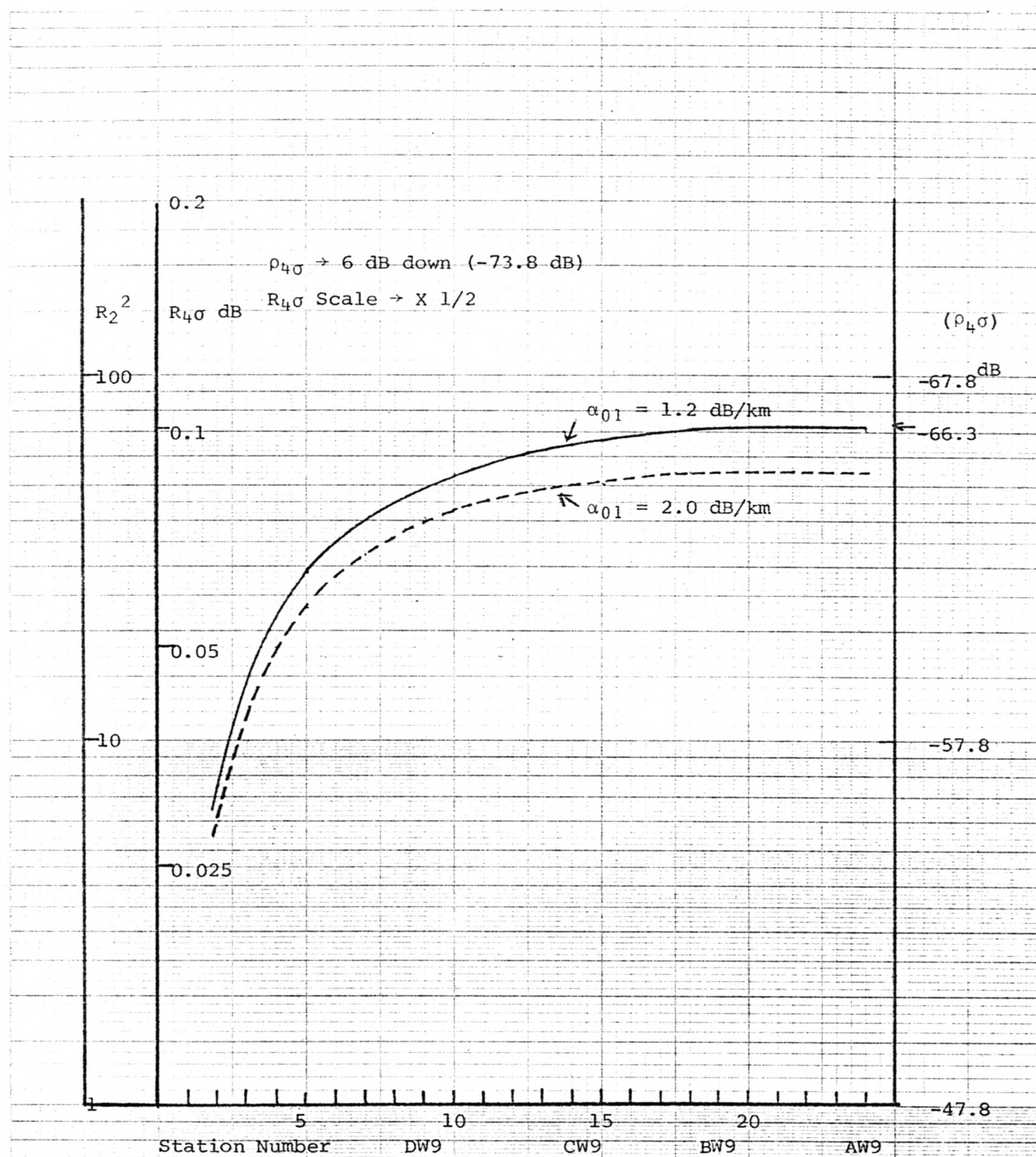
Requirement for getting  $4\sigma_2 \leq 0.1 \text{ dB}$  at 24th station under  $\alpha_{01} = 1.2 \text{ dB/km}$  is

$$\zeta_{f0} \cdot \zeta_{b0} \leq 4.80 \times 10^{-4} \quad (-66.4 \text{ dB})$$

Requirement for getting  $4\sigma_3 \leq 0.1 \text{ dB}$  at the first station under  $\alpha_{01} = 1.2 \text{ dB/km}$  is

$$\zeta_{f0} \cdot D_0 \leq 1.42 \times 10^{-3} \quad (-56.9 \text{ dB})$$

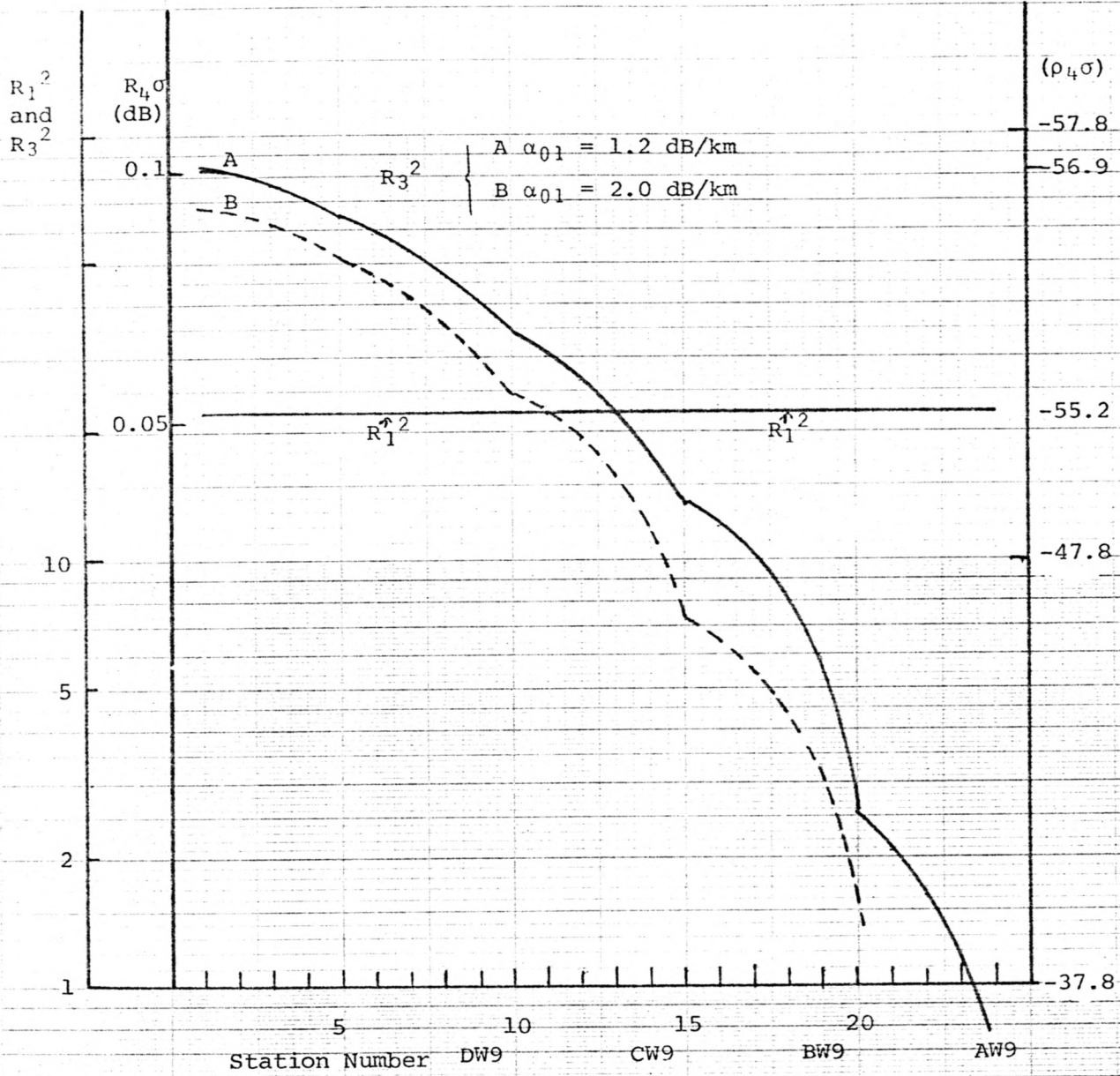
Requirements at other stations are listed in Table 9-1.



$\rho_{4\sigma}$ ; the requirement for  $\zeta_b \cdot \zeta_{f0}$  to get  $4\sigma_2 \leq .1$  dB  
 $R_{4\sigma}$ ;  $4\sigma$  (dB) under  $\rho_{4\sigma} = -66.3$  dB at 24th station  
 $R_2^2 = \sum_{j=1}^{i-1} \sum_{k=j+1}^{N_s} \{ e^{-2\alpha_{01}(l_k - l_j) - 2\alpha_k j} \}^2$       $i$ ; station number  
 $\sigma_2^2 = \frac{1}{2} (\zeta_b \zeta_{f0})^2 R_2^2$   
 $R_{4\sigma} = 17.3 \times 2\sigma$

FIGURE 9-2: TE<sub>01</sub> MODE MULTI REFLECTION EFFECT BETWEEN COUPLERS





$\rho_4\sigma$ ; the requirement for  $\zeta_s \zeta_{fo}$  and  $D_o \zeta_{fo}$  to get  
 $4\sigma_1 \leq 0.1$  dB and  $4\sigma_3 \leq 0.1$  dB

$R_4\sigma$ ;  $4\sigma$  (dB) under  $\rho_4\sigma = -56.9$  dB

FIGURE 9-3: TE<sub>01</sub> MODE MULTI REFLECTION EFFECT BETWEEN COUPLERS AND SIGNAL DISTRIBUTOR AND REVERSE COUPLING

Station No.	1	2		5		10		15		20		23	24
Signal Distributor - coupler front	-55.2 dB (-75.0)	-55.2		-55.2		-55.2		-55.2		-55.2		-55.2 (-75.0)	-55.2
Coupler Back - coupler front	-56.6 dB (-85.0)	-56.2		-62.3		-65.1		-66.0		-66.3		-66.3 (-85.0)	-66.3
Directivity - coupler front	-56.9 dB (-55.0)			-55.9		-53.3		-49.1		-41.8		-44.0 (-70.0)	--
TE <sub>02</sub> Generation	--	-22.0 dB		-28.7 (-32.2)		-31.6 (-38.3)		-32.9 (-41.4)		-33.3		-33.4	-33.4 (43.1)
TE <sub>02</sub> Coupling													

( ) ; requirement to suppress variation less than 0.1 dB<sup>P-P</sup>

TABLE 9-1: REQUIREMENT FOR SPURIOUS SIGNAL LEVEL TO GET  $4\sigma < 0.1$  dB

## 10. SPECIFICATION FOR COUPLERS

Almost all the ripple mechanisms in VLA Waveguide System has been reviewed in this report. In addition there are some waveguide components, such as couplers, adapters, and signal distributors, which cause some variation in attenuation response, but usually their frequency response is not so fast nor so complicated as the  $TE_{01}$  mode reflection effect or  $TE_{on}$  mode conversion effect. So when the performance in the narrow band of 10 MHz or so is considered, only the latter should be considered. Those are listed again below:

[Main Waveguide]

1.  $TE_{01}$  mode reflection effect caused by couplers
2.  $TE_{on}$  mode conversion effect

[Coupler]

3.  $TE_{on}$  mode coupling effect

[Antenna Waveguide]

4.  $TE_{01}$  mode reflection effect between adapter and coupler

Item 4 can be considered to be identical for all antenna stations and the effect of Item 3 becomes smaller at stations far away from the center. From those facts, it would seem reasonable to have the following assignment of variation to each factor to get the design goal of less than 0.2 dB between  $2\sigma$  values in 10 MHz band through all the system.

[Stations near the center]

1. } less than 0.14 dB
2. }
3. less than 0.10 dB
4. less than 0.10 dB

[Stations far away from the center]

1. } less than 0.17 dB
2. }

3. less than 0.05 dB
4. less than 0.10 dB

To get these values, the design goal of 90° sector coupler that seems to be the most promising and to be used at almost twenty stations should be set as shown in Table 10-1. Some more basic data to get this design goal is listed in Table 10-2.

<Main Line>	<Design Goal>	<Present Situation>
1. Insertion loss	less than 0.125 dB (in ch 7 ~ 11)	≈ .15 dB
2. Return loss front	greater than 47 dB	(40 ~ 50 dB)
back	greater than 20 dB	ok (≈ 40 dB)
3. TE <sub>02</sub> mode generation	less than -33 dB	ok
<Coupling>		
1. Coupling value	-20 ~ -30 dB	ok (nominal -25 dB)
2. Coupling variation	less than 1.0 dB in any channel	ok
	less than 0.5 dB in more than 6 channels	ok
3. Return loss	greater than 20 dB	ok
3 TE <sub>02</sub> discrimination	greater than 20 dB	13 ~ 14 dB
<Directivity>	greater than 10 dB	ok (10 ~ 13 dB)

TABLE 10-1: DESIGN GOAL OF 9° Sector Coupler

Station No.	1	2	5	10	15	20	23	24
I. TE <sub>01</sub> Mode Reflection Effect								
A. Signal Distributor - coupler front	-55.2 dB (-75.0)	-55.2	-55.2	-55.2	-55.2	-55.2	-55.2 (-75.0)	-55.2
B. Coupler Back - coupler front	-56.6 dB (-85.0)	-56.2	-62.3	-65.1	-66.0	-66.3	-66.3 (-85.0)	-66.3
C. Directivity - coupler front	-56.9 dB (-55.0)		-55.9	-53.3	-49.1	-41.8	-44.0 (-70.0)	--
II. TE <sub>02</sub> Mode Effect TE <sub>02</sub> Mode Conversion Effect	--	-22.0 dB	-28.7 (-32.2)	-31.6 (-38.3)	-32.9 (-41.4)	-33.3	-33.4	-33.4 (43.1)
TE <sub>02</sub> Coupling								

( ); requirement to suppress variation less than 0.1 dB<sup>P-P</sup>

TABLE 10-2: REQUIREMENT FOR SPURIOUS SIGNAL LEVEL TO GET  $4\sigma \leq 0.1$  dB

11. SUMMARY OF ANALYSIS AND CONCLUSIONS

If there exist two mode conversion sources with the distance  $l_c$ , the ripple of the following properties occurs:

(ripple period)

$$f_{px} = \frac{2\pi}{B_x l_c} \quad (11-1)$$

(ripple amplitude)

$$R_{max} = 17.3 C_c \cdot C_r \cdot e^{-\Delta\alpha l_c} \text{ (dB}^{P-P}) \quad (11-2)$$

(effect to L.O. system)

$$\Delta\phi_c = C_c \cdot C_r \cdot e^{-\Delta\alpha l_c} \cdot 2 \cos \frac{(\Delta\beta + \Delta\beta') l}{2} \sin \frac{(\Delta\beta - \Delta\beta') l}{2} \quad (11-3)$$

At that time when the distance  $l_c$  changes uniformly according to the waveguide line the following formula by  $\Delta l_c$ :

$$\Delta l = t \cdot l \quad (11-4)$$

t: constant (is determined by temperature change or other factors around waveguide and expected to be less than  $10^{-4}$  in main 60 mm waveguide line)

The phase difference between two pilot signals is affected by this mode conversion by the following amount:

$$\begin{aligned} \Delta\Delta\phi_c &\leq C_c C_r e^{-\Delta\alpha l_c} \beta_x \cdot \Delta f_p \cdot \Delta l_c \\ &= \frac{R_{max}^{P-P}}{17.3} \cdot \frac{f_p}{f_{px}} \cdot t \cdot 2\pi \text{ (rad)} \\ \text{(optimized case; } \frac{(\Delta\beta - \Delta\beta') l}{2} &= \frac{\Delta f_p}{f_{px}} \cdot \pi = m\pi) \end{aligned} \quad (11-5)$$

$$\begin{aligned}
\Delta\Delta\phi_c &\leq C_c C_r e^{-\Delta\alpha l_c} \cdot 2\Delta\beta\Delta l_c & (11-6) \\
&= \frac{R_{\max}^{P-P}}{17.3} \cdot \frac{l_c}{\lambda_b} \cdot t \cdot 2\pi \text{ (rad)} \\
\text{(worst case; } &\frac{(\Delta\beta - \Delta\beta')l}{2} = \frac{\Delta f_p}{f_{px}} \cdot \pi = \\
&\left. (m+\frac{1}{2})\pi \text{ and } \Delta l_c \ll \frac{\pi}{2\Delta\beta} = \frac{\lambda_b}{4} \right\}
\end{aligned}$$

If there exist two reflection sources with distance  $l_\zeta$ , the ripple of the following properties occurs:

(ripple period)

$$f_p = \frac{2\pi}{2 \frac{d\beta}{df} \cdot l_\zeta} \doteq \frac{150}{l_\zeta} \text{ (MHz, } l_i \text{ in m)} \quad (11-7)$$

(ripple amplitude)

$$R_{\max}^{P-P} = 17.3 \zeta_s \zeta_s' e^{-2\alpha_{01} l_s} \text{ (dB}^{P-P}) \quad (11-8)$$

(effect to L.O. system)

$$\Delta\phi_\zeta = \zeta_s \cdot \zeta_s' e^{-2\alpha_{01} l_\zeta} \cdot 2 \cos(\beta_{01} + \beta_{01}') l \cdot \sin(\beta_{01} - \beta_{01}') l \quad (11-9)$$

When the distance  $l_\zeta$  changes uniformly according Formula 11-4 the phase difference change between two pilot signals can be shown as follows:

$$\begin{aligned}
\Delta\Delta\phi_\zeta &= \zeta_s \cdot \zeta_s' e^{-2\alpha_{01} l_\zeta} \cdot 2 \cdot (\beta_{01} - \beta_{01}') \Delta l_\zeta & (11-10) \\
&= \frac{R_{\max}^{P-P}}{17.3} \cdot \frac{\Delta f_p}{f_p} \cdot t \cdot 2\pi \text{ (rad)} \\
&\text{(optimized case; } (\beta_{01} - \beta_{01}') l = m\pi \text{)}
\end{aligned}$$



$$\begin{aligned}
&= 4 \cdot \zeta_s \cdot \zeta_s \cdot e^{-2\alpha_{01} \ell \zeta} \\
&= \frac{4 \cdot R_{\max}^{P-P}}{17.3} \quad (\text{rad})
\end{aligned}
\tag{11-11}$$

{worst case;  $(\beta_{01} - \beta_{01}') \ell = (m + \frac{1}{2})\pi$  and

$$\Delta \ell_{\zeta} > \frac{\pi}{\beta_1 + \beta_1'} \doteq \frac{\lambda}{4}$$

From those analysis, the following can be concluded:

- 1) Usually, the ripple caused by  $TE_{01}$  mode reflections has finer ripple and much more unstable than the one caused by mode conversions.
- 2) By suitable channel assignment to each station, the instability of the L.O. system caused by mode conversion could be minimized. (Especially 5 MHz pilot signals.)
- 3) The ripple caused by coupler and adapter reflections could have worst effect to the L.O. system and would be most unstable ripple.
- 4) By having a fixed attenuation at suitable positions in the antenna waveguide route, the instability of the L.O. system caused by  $TE_{01}$  mode reflection effect between coupler and adapter could be minimized.
- 5) The effect of mode conversions or  $TE_{01}$  mode conversions on to the L.O. system could be bigger than that of the first order delay.
- 6) The ripples caused by mode conversions become less unstable in higher frequency than lower frequency. But the ones caused by  $TE_{01}$  mode reflection become more unstable in higher frequency than lower frequency.

APPENDIX A. THE EFFECT OF 1-100 KHZ OFFSET IN LO SYSTEM

The correction error caused by an offset in the LO round-trip measuring system can be calculated as follows: (refer to Formulae 3-7 and 3-8)

$$\begin{aligned}\phi_e &= \phi_t - \phi_r \\ &= -(\beta_{01} - \beta_{01}') \cdot \ell + C_c C_r e^{-\Delta\alpha \cdot \ell} \sin \Delta\beta \ell - C_c' \cdot C_r' e^{-\Delta\alpha' \ell} \sin \Delta\beta' \ell\end{aligned}\quad (1)$$

$\phi_e$ ; correction error

$\phi_t$ ; phase shift of  $TE_{01}$  mode at  $f_p$

$\phi_r$ ; phase shift of  $TE_{01}$  mode at  $f_p + \Delta f_p$  (offset frequency)

As  $\Delta f_p$  is small enough to assume  $C_c = C_c'$ ,  $C_r = C_r'$ ,  $\Delta\alpha = \Delta\alpha'$   $\phi_e$  becomes:

$$\begin{aligned}\phi_e &= \phi_{e\ell} + \phi_{ec} \\ &= -(\beta_{01} - \beta_{01}') \cdot \ell + C_c \cdot C_r e^{-\Delta\alpha \ell} (\sin' \Delta\beta \ell - \sin \Delta\beta' \ell)\end{aligned}\quad (2)$$

$\phi_{e\ell}$ ; correction error caused by the first order delay

$\phi_{ec}$ ; correction error caused by mode conversion

The first term in (2) can be calculated as follows:

$$\begin{aligned}\phi_{e\ell} &= -(\beta_{01} - \beta_{01}') \ell \\ &= \frac{2\pi}{300} \cdot \frac{1}{\sqrt{1 - (f_{c01}/f)^2}} \cdot \Delta f_p \cdot \ell \\ &\doteq \frac{2\pi}{300} \cdot \Delta f_p \cdot \ell\end{aligned}\quad (3)$$

At the farthest station that is 21 Km away from the center,

$$\phi_{e\ell} \doteq 25.5 \text{ (deg) for } \Delta f_p = 1 \text{ KHz}$$

$$\doteq 2550 \text{ (deg) for } \Delta f_p = 100 \text{ KHz}$$

The second term in (2) can be calculated as follows:

$$\begin{aligned} \phi_{ec} &= C_c \cdot C_r \cdot e^{-\Delta\alpha\ell} (\sin\Delta\beta\ell - \sin\Delta\beta'\ell) \\ &= C_c \cdot C_r \cdot e^{-\Delta\alpha\ell} (\Delta\beta - \Delta\beta') \cdot \ell \\ &= C_c \cdot C_r \cdot e^{-\Delta\alpha\ell} \beta_x \cdot \Delta f_p \cdot \ell \end{aligned} \quad (4)$$

When TE<sub>02</sub> mode is considered as spurious mode at 35 GHz:

$$\begin{aligned} \phi_{ec} &= 4.68 \times 10^{-2} \times \ell \cdot C_c C_r e^{-\Delta\alpha\ell} \cdot \text{(deg)} \text{ (for } \Delta f_p = 1 \text{ KHz)} \\ &= 4.68 \times \ell \cdot C_c \cdot C_r e^{-\Delta\alpha\ell} \text{ (deg)} \text{ (for } \Delta f_p = 100 \text{ KHz)} \end{aligned} \quad (5)$$

( $\ell$  in km)

The correction error caused by an offset of 1 - 100 KHz goes up more than 25 degrees easily at the farthest antenna and cannot be neglected. Such error mostly depends upon the first order delay and is determined by the distance between transmitter and receiver. And it should be stable to temperature change. So once it is known, it could be corrected by another process.

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