

140-Foot Telescope Pointing Calibration

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Summary

Expressions for the pointing corrections necessary to compensate for encoder angular offsets, beam collimation errors, axis perpendicularity errors, polar axis orientation errors, reflector and yoke flexures, encoder eccentricities, and cyclic errors are derived.

The accuracy with which corrections must be computed, and specially the influence of atmospheric conditions, are discussed.

Optimum procedures for single source observations are analyzed, and an example of gaussian beam fitting is given.

A tentative list of calibration sources is included with some discussion on the overall observing procedure.

A performance index is defined and applied to the determination of the optimum observing wavelength. A programming example of pointing correction parameter fitting is included with an application to data prepared from the standard pointing correction curves, which are also reproduced.

Expressions for structural and thermal focal point changes are derived and discussed.

Definition of rectangular coordinate systems

Four cartesian coordinate systems will be defined. The axis orientations have been chosen so that the number of rotation matrices necessary to perform transformations between systems is minimized, while reasonably intuitive designations are preserved. Other choices to suit different tastes are possible without altering in substance the resulting pointing equations.

Astrometric system

- X₁ positive towards the source
- Y₁ on the hour angle semiplane through the source, oriented positive towards increasing declinations
- Z₁ perpendicular to the hour angle semiplane through the source, oriented positive towards increasing hour angles.

Intermediate equatorial system

- X₂ positive towards the intersection of the hour angle semiplane through the source with the equator
- Y₂ positive towards the north pole
- Z₂ identical with Z₁

Equatorial system

- X₃ positive towards the intersection of the meridian elevated semiplane with the equator
- Y₃ identical with Y₂
- Z₃ positive towards the west cardinal point

Altazimuth system

- X₄ positive towards the south cardinal point
- Y₄ positive towards the zenith
- Z₄ identical with Z₃

Abbreviations

- D declination
- H hour angle
- L latitude
- P position angle
- ΔD error in declination
- ΔH error in hour angle
- ΔP error in position angle
- s sine
- c cosine
- t tangent

Conversion between systems

Transformations are given in one sense only. To express the inverse transformation note that the inverse of a rotation matrix is equal to its transpose (because rotation matrices are orthonormal).

Astrometric to intermediate equatorial

Rotation of $-D$ in the counterclockwise sense around the Z_1 axis

$$\begin{bmatrix} X_2 \\ Y_2 \\ Z_2 \end{bmatrix} = \begin{bmatrix} cD & -sD & 0 \\ sD & cD & 0 \\ 0 & 0 & 1 \end{bmatrix} \cdot \begin{bmatrix} X_1 \\ Y_1 \\ Z_1 \end{bmatrix}$$

Intermediate equatorial to equatorial

Counterclockwise rotation of H around the Y_2 axis

$$\begin{bmatrix} X_3 \\ Y_3 \\ Z_3 \end{bmatrix} = \begin{bmatrix} cH & 0 & -sH \\ 0 & 1 & 0 \\ sH & 0 & cH \end{bmatrix} \cdot \begin{bmatrix} X_2 \\ Y_2 \\ Z_2 \end{bmatrix}$$

Equatorial to altazimuth

Counterclockwise rotation of $-(90-L)$ around the Z_3 axis

$$\begin{bmatrix} X_4 \\ Y_4 \\ Z_4 \end{bmatrix} = \begin{bmatrix} sL & -cL & 0 \\ cL & sL & 0 \\ 0 & 0 & 1 \end{bmatrix} \cdot \begin{bmatrix} X_3 \\ Y_3 \\ Z_3 \end{bmatrix}$$

Derivation of error terms

Notice that the various error constants that appear may be often given an arbitrary sign. A basic physical interpretation for an error is given, but other error terms may have the same functional form and will eventually be added together in a single term for computation. The hour angle errors are better expressed as ΔH , cD , the angular error in the sky (error along the Z_1 axis of the astrometric system).

Index error in declination

Corresponds to an arbitrary rotation of the encoder relative to the shaft on which it is mounted.

$$\Delta D = A_1$$

$$\Delta H.cD = 0$$

Index error in hour angle

Similar in concept to the declination term

$$\Delta D = 0$$

$$\Delta H.cD = A_2.cD$$

Collimation error

Due to the error in perpendicularity between the antenna beam axis and the declination axis. The angular error in hour angle $\Delta H.cD$ is constant and equal to the collimation error

$$\Delta D = 0$$

$$\Delta H.cD = A_3$$

Perpendicularity error

Caused by the error in perpendicularity between the declination axis and the hour angle axis. Consider a perfect telescope in which the perpendicularity error is zero. We may transform the beam of this ideal telescope into the beam of the real telescope by a rotation vector of magnitude equal to the perpendicularity error angle, oriented along X_2 in the intermediate equatorial system. The components of this vector in the astrometric system are the pointing errors (ΔP , $\Delta H.cD$, ΔD) and may be computed as follows

$$\begin{bmatrix} \Delta P \\ \Delta H.cD \\ \Delta D \end{bmatrix} = \begin{bmatrix} cD & sD & 0 \\ -sD & cD & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} A_4 \\ 0 \\ 0 \end{bmatrix}$$

so that

$$\Delta D = 0$$

$$\Delta H.cD = -sD.A_4$$

where the minus sign may be absorbed in the arbitrary constant A_4

Polar axis orientation errors

These are due to the polar axis not pointing precisely to the true celestial pole. As indicated for the preceding case, we transform between the ideal beam and the actual beam by applying a suitable rotation vector. In this case the equatorial system is the most convenient and the rotation vector is represented in it by $(A_5, 0, A_6)$. Transforming this vector into the astrometric system we obtain:

$$\begin{bmatrix} \Delta P \\ \Delta H \text{ cD} \\ \Delta D \end{bmatrix} = \begin{bmatrix} \text{cH cD} & \text{sD} & \text{sH cD} \\ -\text{cH sD} & \text{cD} & -\text{sH sD} \\ -\text{sH} & 0 & \text{cH} \end{bmatrix} \begin{bmatrix} A_5 \\ 0 \\ A_6 \end{bmatrix}$$

and

$$\begin{aligned} \Delta D &= -\text{sH} A_5 + \text{cH} A_6 \\ \Delta H \text{ cD} &= -\text{cHsD} A_5 - \text{sHsD} A_6 \end{aligned}$$

Refraction

Ilif and Holt (Journal of Research of the National Bureau of Standards, 67 D, 31, 1963) have investigated atmospheric refraction at 1.9 cm wavelength with a radio sextant tracking the sun. They arrive at the following empirical expression for the refraction angle r in degrees:

$$r = \left(\frac{180}{\pi} \times 10^{-6} \right) \left[\cot h - \frac{D}{(h+E)^F} \right] N_s - \frac{A}{(h+B)C}$$

where h is the altitude in degrees, A, B, C, D, E, F are positive empirical constants, and N_s is the surface refractivity ($N_s = (n - 1) \times 10^6$, n index of refraction at the surface).

Substituting typical values

$$\begin{aligned} r &= \left(\frac{180}{\pi} \times 10^{-6} \right) \left[\cot h - \frac{43}{(h+0.4)^{2.64}} \right] \cdot 325 \dots \\ &\quad - \frac{40}{(h+2.7)^4} \end{aligned}$$

The second and third terms in this formula are small correction terms. For the example given above we have

Elevation	2nd and 3rd Terms
20°	- 1."54
30°	- 0."47
40°	- 0."15

We may neglect these terms for practical purposes although there would be no difficulty in including them if it were necessary.

To a very good approximation assume then the refraction angle proportional to the tangent of the zenith distance. The direction of the zenith is given in the altazimuth system by the unit vector (0, 1, 0).

Transforming this vector to the astrometric system we have

$$\begin{bmatrix} X_z \\ Y_z \\ Z_z \end{bmatrix} = \begin{bmatrix} sL cHcD - cLsD & -sLcHsD - cLcD & -sLsH \\ cL cHcD + sLsD & -cLcHsD + sLcD & -cLsH \\ sH sD & -sHsD & cH \end{bmatrix} \cdot \begin{bmatrix} 0 \\ 1 \\ 0 \end{bmatrix}$$

$$\begin{bmatrix} X_z \\ Y_z \\ Z_z \end{bmatrix} = \begin{bmatrix} cL cH cD + sL sD \\ -cL cH sD + sL cD \\ -cL sH \end{bmatrix}$$

Refraction displaces the antenna beam towards the zenith by an amount proportional to the tangent of the zenith distance (tZD)

$$tZD = \frac{\frac{\sqrt{Y_z^2 + Z_z^2}}{Z_z}}{X_z}$$

and corresponding errors in declination and hour angle are given by

$$\Delta D = A_7 \cdot tZD \cdot \frac{Y_z}{\sqrt{Y_z^2 + Z_z^2}}$$

$$\Delta H.cD = A_7 \cdot tZD \cdot \frac{Z_z}{\sqrt{Y_z^2 + Z_z^2}}$$

or substituting

$$\Delta D = A_7 \cdot \frac{-cL \ cH \ sD + sL \ cD}{cL \ cH \ cD + sL \ sD}$$

$$\Delta H \ cD = A_7 \frac{-cL \ sH}{cL \ cH \ cD + sL \ sD}$$

Reflector flexure

We assume that the North-South bending is proportional to the North-South component of gravity, and similarly for the East-West bending, for any orientation. These assumptions should be good if the structure is not suffering from significant buckling or other structural hysteresis effects.

Because of the symmetry of the structure (the asymmetry introduced by the declination drive is negligible with good balancing) the N-S pointing error is due exclusively to the N-S component of the gravitational loading (and similarly for the E-W pointing error).

We have already computed the direction cosines for the zenith in connection with the refraction section above. Therefore we may write directly

$$\Delta D = A_8 (-cL \ cH \ sD + sL \ cD)$$

$$\Delta H \ cD = A_9 (-cL \ sH)$$

Fork flexure

We use arguments analogous to those employed in the case of the reflector flexure. The components of the unit vector directed to the nadir are (0, -1, 0) in the altazimuth system, and are given in the intermediate equatorial system by

$$\begin{pmatrix} X_2 \\ Y_2 \\ Z_2 \end{pmatrix} = \begin{pmatrix} sLcH & cLcH & sH \\ -cL & sL & 0 \\ -sLsH & -cLsH & cH \end{pmatrix} \begin{pmatrix} 0 \\ -1 \\ 0 \end{pmatrix}$$

$$\begin{bmatrix} X_2 \\ Y_2 \\ Z_2 \end{bmatrix} = \begin{bmatrix} cL \ cH \\ sL \\ -cL \ sH \end{bmatrix}$$

Consider now how the distortions of the yoke affect the pointing of the reflector. The reflector is structurally connected to the yoke through the declination bearings and the declination drive. These three points determine a plane. Changes in the orientation of this plane will produce pointing errors. The pointing errors can be conveniently represented by a rotation vector with components (X_r, Y_r, Z_r) in the intermediate equatorial system. Since the yoke is symmetric with respect to the X_2, Y_2 plane, the X_2 and Y_2 components of gravity will contribute only to the Z_r component. The Z_2 component of gravity (antisymmetric with respect to the plane of symmetry) contributes only to the X_r and Y_r components. We have then:

$$\begin{bmatrix} X_r \\ Y_r \\ Z_r \end{bmatrix} = \begin{bmatrix} A_{10} \ sH \\ A_{11} \ sH \\ A_{12} \ cH \end{bmatrix}$$

The latitude factors and signs have been absorbed in the arbitrary constants.

Transforming these vectors to the astrometric system, the pointing errors become:

$$\begin{bmatrix} \Delta P \\ \Delta HcD \\ \Delta D \end{bmatrix} = \begin{bmatrix} cD \ sD \ 0 \\ -sD \ cD \ 0 \\ 0 \ 0 \ 1 \end{bmatrix} \cdot \begin{bmatrix} A_{10} \ sH \\ A_{11} \ sH \\ A_{12} \ cH \end{bmatrix}$$

and

$$\Delta D = A_{12} \ cH$$

$$\Delta HcD = -A_{10} \ sD \ sH + A_{11} \ cD \ sH$$

Encoder eccentricity errors

Due to eccentricity between the rotor and stator windings of the inductosyns. The error is sinusoidal with the angle.

$$\Delta D = A_{13} \ sD + A_{14} \ cD$$

$$\Delta HcD = A_{15} \ sH \ cD + A_{16} \ cH \ cD$$

Cyclic errors

Due to the intrinsic encoder cyclic error with a period equal to the angular spacing of the inductosyn poles (1° in our case). This error always exists in some small amount and can be magnified by any significant unbalance between the excitations of the sine and cosine windings.

$$\Delta D = A_{17} \sin \frac{180}{\pi} D + A_{18} \cos \frac{180}{\pi} D$$

$$\Delta HcD = A_{19} \sin \frac{180}{\pi} H \cdot cD + A_{20} \cos \frac{180}{\pi} H \cdot cD$$

Summary of pointing corrections

Let us summarize now all the correction terms described above, consolidating all terms of the same functional form, and renumbering sequentially the arbitrary constants. Note that the "empirical terms" in Pauliny Toth's memo of May 7, 1969 are all recovered in this treatment.

<u>Declination terms</u>	$\Delta D = \sum$	
A_1		Indexing
$A_2 \text{ sH} + A_3 \text{ cH}$		Polar axis orientation, Yoke flexure
$A_4 \frac{-cL \text{ cH} \text{ sD} + sL \text{ cD}}{cL \text{ cH} \text{ cD} + sL \text{ sD}}$		Refraction
$A_5 \text{ cH} \text{ sD}$		Reflector flexure
$A_6 \text{ sD} + A_7 \text{ cD}$		Eccentricity, reflector flexure
$A_8 \sin \frac{180}{\pi} D + A_9 \cos \frac{180}{\pi} D$		Cyclic error

Hour angle terms $\Delta H \text{ cD} = \sum$

$A_{10} \text{ cD}$	Indexing
A_{11}	Collimation
$A_{12} \text{ sD}$	Perpendicularity
$A_2 \text{ cH sD} + A_{14} \text{ sH sD}$	Polar axis orientation, Yoke flexure
$A_4 \frac{-\text{cL sH}}{\text{cL cH cD} + \text{sL sD}}$	Refraction
$A_{15} \text{ sH}$	Reflector flexure
$A_{16} \text{ cD cH} + A_{17} \text{ cD sH}$	Yoke flexure, eccentricity
$A_{18} \sin \frac{180}{\pi} H \cdot \text{cD} + A_{19} \cos \frac{180}{\pi} H \cdot \text{cD}$	Cyclic error

Accuracy in the computation of error terms

If we aim for an rms pointing accuracy of 1/10 of the beamwidth at 2 cm, and allow for a large number of random contributions to the pointing error budget (say 16), single contributions should be computed with an rms accuracy of 3 arcseconds.

With this criterion, some error terms may be neglected. Residual periodic errors, in the inductosyn angular encoders, are expected to be under 3 seconds. However, because of difficulties in accurately balancing the drives to the quadrature windings for various temperatures and over a long period of time, it is desirable to preserve a capability to search for periodic errors. It is desirable to have integration times of a fraction of the time required to vary the hour angle by 1° (4 minutes of time), at least for a subset of the observations.

Influence of atmospheric conditions

The angle of refraction is affected by atmospheric temperature, pressure, and humidity. Froome and Essen (The Velocity of Light and Radio Waves, Academic Press 1969, p. 24) give expressions for the index of refraction as a function of atmospheric conditions. A simplified formula (but quite accurate for ordinary conditions and frequencies below 30 GHz) is the following

$$N = (n-1) \times 10^6 = \frac{103 P_1}{T} + \frac{86}{T} \left(1 + \frac{5750}{T}\right) P_2$$

where T is the absolute temperature, P_1 is the partial pressure of dry air, and P_2 is the partial pressure of water vapor (expressed in mm of mercury).

For a total pressure of 760 mm, 20°C temperature, and 50% relative humidity, (8.9 mm water vapor pressure, refer to psychrometric chart).

$$\frac{dN}{dp} = 0.35$$

where p is the total pressure

$$\frac{dN}{dp_2} = 6.1$$

$$\frac{dN}{dT} = -1.3$$

Under these conditions the refractivity is equal to 320. The refraction angle is approximately given by

$$r = (n-1) \cdot \text{tg } z$$

and its derivatives are

$$\frac{dr}{dp} = 0.07 \text{ tg } z \text{ arcseconds/mm total pressure.}$$

$$\frac{dr}{dp_2} = 1.28 \text{ tg } z \text{ arcseconds/mm water vapor pressure}$$

$$\frac{dr}{dT} = 0.26 \text{ tg } z \text{ arcseconds/}^\circ\text{C}$$

For a 3" error in the refraction angle, at altitudes of not less than 30°, we can tolerate errors of 25 mm in the total pressure, 1.3 mm in the water vapor pressure, and 6.7°C in the temperature.

Referring to the psychrometric chart, the wet minus dry bulb temperature difference has to be known with an accuracy of about 2°F or 10% in the relative humidity.

2-30 LOAD CALCULATIONS

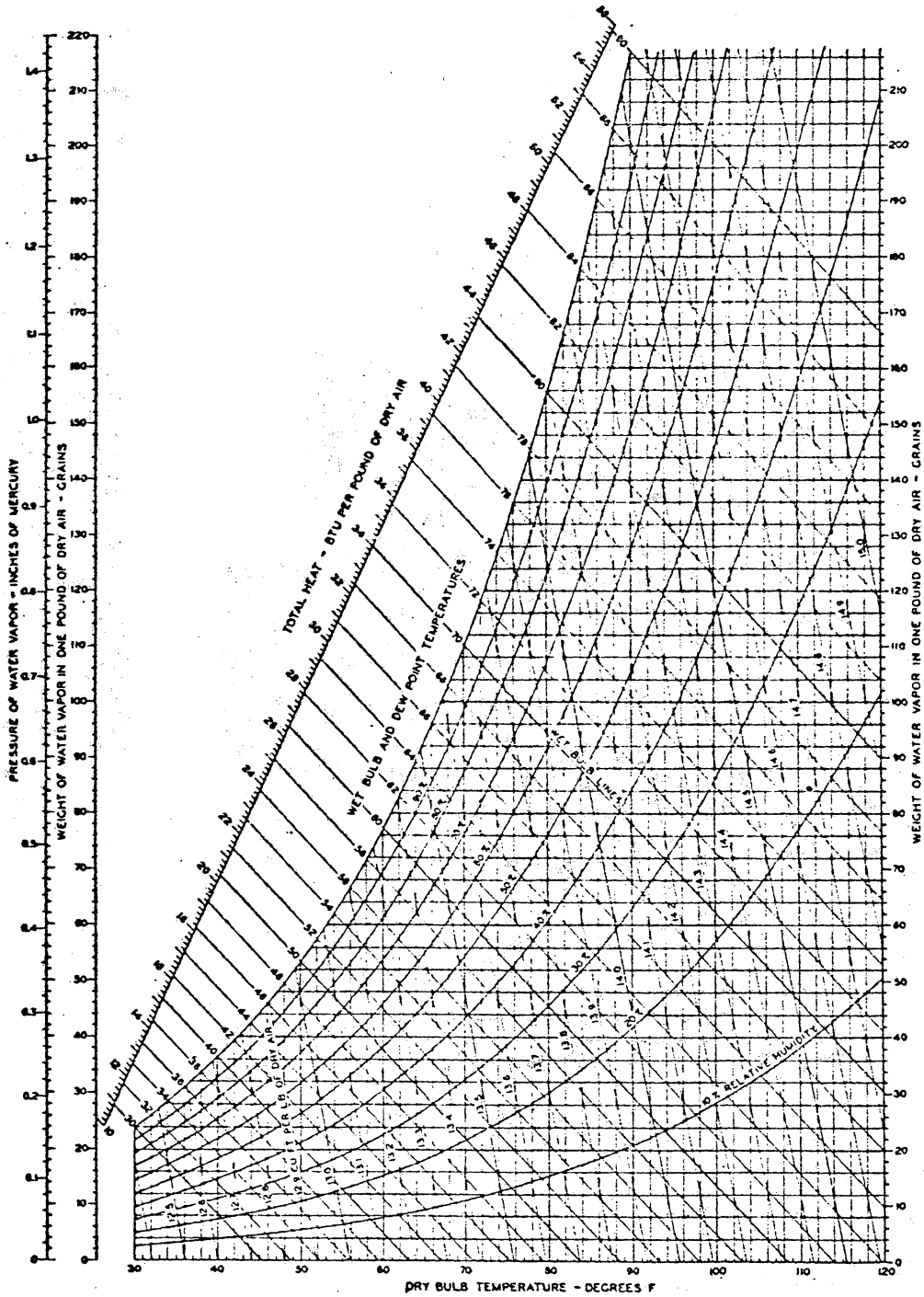


Fig. 1. Psychrometric properties of air at 29.92 inches of mercury absolute pressure.

Observing strategy for single sources

The shape of the main antenna beam can be closely approximated by a Gaussian function of two variables.

For our application, the following are a sufficient number of parameters, to be determined from observations:

- a) amplitude of the radio source emission
- b) antenna beam width, beam assumed centrally symmetric
- c) angular offsets of the beam; these are the pointing errors of main interest
- d) baseline level at the beam center
- e) baseline slopes in declination and hour angle

The antenna beam width should be retained as a free parameter since gravitational and other distortions will change its value with orientation and time. Baseline slopes may be significant because of ground noise pickup through the sidelobes.

We may derive some basic useful relations by considering a one dimensional Gaussian function as follows

$$y = a + b e^{-\frac{x^2}{2\sigma^2}}$$

- where
- y - receiver output
 - a - baseline level
 - b - source amplitude
 - σ - standard deviation
(HPBW = 2.35 σ)
 - x - angle between the source and the beam center

then

$$\frac{dy}{dx} = -\frac{x}{\sigma^2} y$$

$$\frac{d^2y}{dx^2} = \left(\frac{x^2}{\sigma^4} - \frac{1}{\sigma^2} \right) y$$

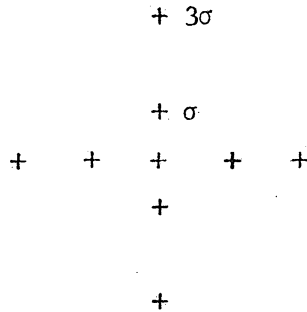
$$\frac{dy}{d\sigma} = \frac{x^2}{\sigma^3} y$$

$$\frac{d^2y}{d\sigma^2} = \left(\frac{x^4}{\sigma^6} - \frac{3x^2}{\sigma^4} \right) y$$

From the preceding, it may be concluded that the receiver output displays maximum sensitivity to variations of the angular offsets for $x = \sigma$, and to variations of the beamwidth for $x = \sqrt{3} \sigma$

A good determination of the baseline levels and slopes should be made from measurements far from the beam center. An angular offset equal to 1% of the beamwidth (3 arcseconds at 6 cm) will produce output changes of the order of magnitude of 1% of the peak response. For angles of more than 3σ , the amplitude response of the main beam becomes much less than 1%, and this is the minimum distance for good baseline determinations.

Based on the preceding criteria, many observing arrangements are possible. Take as a good example the one represented by the following diagram where the crosses indicate angular offsets from the nominal radio source position at the center.



Angular offsets of more than 3σ may be undesirable since distant confusing sources could be picked up. Scanning modes where the telescope continuously moves across the radio source are possible. Servo transients occur near the beginning and end of a scan, and data taking should begin after the servo settles to a steady velocity. This approach seems more complex to program in real time and later analyze.

Crosses with only 4 points symmetrically distributed around the center are undesirable as they lead to ill determined solutions for the angular offsets.

Several schemes have been tried in numerical simulation. The program reproduced below shows one such example for a case of 4 points asymmetrically distributed about the center.

```

C GA
0001      DIMENSION
X OFFSET(2),CR(5),C(50),OFF(10),F(5),
X WORKA(10),WORKB(10),DELP(10)
0002      NRAND=720225
0003      HPBW=0.1
0004      ARAN=0.2
0005      DO 120 NSORC=1,10
0006      CALL RANDU(NRAND,NRAND,FRAND)
0007      SIGMA=HPBW/2.35+(2*FRAND-1)*HPBW*ARAN
0008      CALL RANDU(NRAND,NRAND,FRAND)
0009      HC=FRAND*180-90
0010      CALL RANDU(NRAND,NRAND,FRAND)
0011      DC=FRAND*120-30
0012      CALL RANDU(NRAND,NRAND,FRAND)
0013      HS=HC+(2*FRAND-1)*SIGMA*ARAN/COS(DC)
0014      CALL RANDU(NRAND,NRAND,FRAND)
0015      DS=DC+(2*FRAND-1)*SIGMA*ARAN
0016      AMP=1.+(2*FRAND-1)*ARAN
0017      CALL RANDU(NRAND,NRAND,FRAND)
0018      BASE=(2*FRAND-1)*ARAN
0019      CALL RANDU(NRAND,NRAND,FRAND)
0020      DCE=DC
0021      HCF=HC
0022      BASEE=0
0023      AMPE=1
0024      SIGMAE=HPBW/2.35

C
0025      OFF(1)=0
0026      OFF(2)=0
0027      OFF(3)=HPBW/2
0028      OFF(4)=0
0029      OFF(5)=-OFF(3)
0030      OFF(6)=0
0031      OFF(7)=0
0032      OFF(8)=HPBW/(2*COS(DCE/57.3))
0033      OFF(9)=10*OFF(3)
0034      OFF(10)=0
0035      PRINT 130,BASE,AMP,SIGMA,DS,HS

C
0036      DO 120 NITER=1,10
0037      DO 100 I=1,5
0038      CALL COPY(OFF,I,OFFSET,2,5,0)
0039      C=DC+OFFSET(1)
0040      H=HC+OFFSET(2)
0041      HA=(HCE-H)*COS(D/57.3)
0042      EXPONE=- (HA**2+(DCE-D)**2)/(2*SIGMAE**2)
0043      CR(1)=1
0044      CR(2)=EXP(EXPONE)
0045      CR(3)=-AMPE*CR(2)*(DCE-D)/SIGMAE**2
0046      CR(4)=-AMPE*CR(2)*HA/SIGMAE**2
0047      CR(5)=AMPE*CR(2)*(-2*EXPONE/SIGMAE)
0048      HA=(HS-H)*COS(D/57.3)
0049      EXPON=- (HA**2+(DS-D)**2)/(2*SIGMA**2)
0050      F(I)=BASE+AMP*EXP(EXPON)-BASEE-AMPE*EXP(EXPONE)
0051      DO 110 J=1,5
0052      CALL LOC(I,J,IR,5,5,0)
0053      C(IR)=CR(J)

```



```
0054      110 CONTINUE
0055      100 CONTINUE
C
0056      -CALL MINV(C,5,DET,WORKA,WORKB)
0057      CALL GMRD(C,F,DELP,5,5,1)
0058      BASEE=BASEE+DELP(1)
0059      AMPE=AMPE+DELP(2)
0060      DCE=DCE+DELP(3)
0061      HCE=HCE+DELP(4)
0062      SIGMAE=SIGMAE+DELP(5)
0063      PRINT 130,BASEE,AMPE,SIGMAE,DCE,HCE,NITER,DET
0064      130 FORMAT(IX,5E12.5,15,4E12.5)
0065      120 CONTINUE
C
0066      STOP
0067      END
```

-0.44561E-01	0.11153E	0.61774E-01	0.24747E	02	0.68436E	02	
-0.44561E-01	0.10994E	0.57581E-01	0.24746E	02	0.68434E	02	1-0.62376E 04
-0.44561E-01	0.11142E	0.61491E-01	0.24747E	02	0.68436E	02	2-0.26722E 04
-0.44561E-01	0.11153E	0.61773E-01	0.24747E	02	0.68436E	02	3-0.20127E 04
-0.44561E-01	0.11153E	0.61774E-01	0.24747E	02	0.68436E	02	4-0.19734E 04
0.44561E-01	0.11153E	0.61774E-01	0.24747E	02	0.68436E	02	5-0.19733E 04
-0.44561E-01	0.11153E	0.61774E-01	0.24747E	02	0.68436E	02	6-0.19733E 04
-0.44561E-01	0.11153E	0.61774E-01	0.24747E	02	0.68436E	02	7-0.19733E 04
-0.44561E-01	0.11153E	0.61774E-01	0.24747E	02	0.68436E	02	8-0.19733E 04
-0.44561E-01	0.11153E	0.61774E-01	0.24747E	02	0.68436E	02	9-0.19733E 04
-0.44561E-01	0.11153E	0.61774E-01	0.24747E	02	0.68436E	02	10-0.19733E 04
0.16922E-01	0.93352E	0.59740E-01	0.94982E	00	0.11243E	02	
0.16922E-01	0.93088E	0.54233E-01	0.95112E	00	0.11244E	02	1-0.62372E 04
0.16922E-01	0.93511E	0.59202E-01	0.94988E	00	0.11243E	02	2-0.21273E 04
0.16922E-01	0.93551E	0.59735E-01	0.94982E	00	0.11243E	02	3-0.14368E 04
0.16922E-01	0.93552E	0.59740E-01	0.94982E	00	0.11243E	02	4-0.13774E 04
0.16922E-01	0.93552E	0.59740E-01	0.94982E	00	0.11243E	02	5-0.13768E 04
0.16922E-01	0.93552E	0.59740E-01	0.94982E	00	0.11243E	02	6-0.13769E 04
0.16922E-01	0.93552E	0.59740E-01	0.94982E	00	0.11243E	02	7-0.13768E 04
0.16922E-01	0.93552E	0.59740E-01	0.94982E	00	0.11243E	02	8-0.13768E 04
0.16922E-01	0.93552E	0.59740E-01	0.94982E	00	0.11243E	02	9-0.13768E 04
0.16922E-01	0.93552E	0.59740E-01	0.94982E	00	0.11243E	02	10-0.13768E 04
-0.13161E 00	0.10676E	0.36438E-01	0.85769E	01	0.46633E	02	
-0.13161E 00	0.10578E	0.35459E-01	0.85772E	01	0.46638E	02	1-0.62370E 04
-0.13161E 00	0.10674E	0.36443E-01	0.85769E	01	0.46636E	02	2-0.78030E 04
-0.13161E 00	0.10676E	0.36438E-01	0.85769E	01	0.46636E	02	3-0.80716E 04
-0.13161E 00	0.10676E	0.36436E-01	0.85769E	01	0.46633E	02	4-0.80757E 04
-0.13161E 00	0.10676E	0.36438E-01	0.85769E	01	0.46636E	02	5-0.80757E 04
-0.13161E 00	0.10676E	0.36438E-01	0.85769E	01	0.46638E	02	6-0.80757E 04
-0.13161E 00	0.10676E	0.36438E-01	0.85769E	01	0.46638E	02	7-0.80757E 04
-0.13161E 00	0.10676E	0.36438E-01	0.85769E	01	0.46638E	02	8-0.80757E 04
-0.13161E 00	0.10676E	0.36438E-01	0.85769E	01	0.46638E	02	9-0.80757E 04
-0.13161E 00	0.10676E	0.36438E-01	0.85769E	01	0.46638E	02	10-0.80757E 04
-0.10600E 00	0.10352E	0.42122E-01	0.84766E	02	0.70777E	02	
-0.10600E 00	0.10345E	0.42131E-01	0.84766E	02	0.70780E	02	1-0.62371E 04
-0.10600E 00	0.10352E	0.42122E-01	0.84766E	02	0.70780E	02	2-0.70772E 04
-0.10600E 00	0.10352E	0.42122E-01	0.84766E	02	0.70779E	02	3-0.70909E 04
-0.10600E 00	0.10352E	0.42122E-01	0.84766E	02	0.70779E	02	4-0.70853E 04
-0.10600E 00	0.10352E	0.42122E-01	0.84766E	02	0.70779E	02	5-0.70822E 04
-0.10600E 00	0.10352E	0.42122E-01	0.84766E	02	0.70779E	02	6-0.70783E 04
-0.10600E 00	0.10352E	0.42122E-01	0.84766E	02	0.70778E	02	7-0.70747E 04
-0.10600E 00	0.10352E	0.42122E-01	0.84766E	02	0.70778E	02	8-0.70716E 04
-0.10600E 00	0.10352E	0.42122E-01	0.84766E	02	0.70778E	02	9-0.70609E 04
-0.10600E 00	0.10352E	0.42122E-01	0.84766E	02	0.70778E	02	10-0.70601E 04
-0.14975E 00	0.94630E	0.46254E-01	0.45733E	02	0.10682E	01	
-0.14975E 00	0.91691E	0.45758E-01	0.45733E	02	0.10734E	01	1-0.62370E 04
-0.14975E 00	0.94401E	0.46258E-01	0.45733E	02	0.10690E	01	2-0.43262E 04
-0.14975E 00	0.94613E	0.46254E-01	0.45733E	02	0.10687E	01	3-0.40076E 04
-0.14975E 00	0.94628E	0.46254E-01	0.45733E	02	0.10684E	01	4-0.40510E 04
-0.14975E 00	0.94629E	0.46254E-01	0.45733E	02	0.10683E	01	5-0.40556E 04
-0.14975E 00	0.94630E	0.46254E-01	0.45733E	02	0.10683E	01	6-0.40564E 04
-0.14975E 00	0.94630E	0.46254E-01	0.45733E	02	0.10682E	01	7-0.40566E 04
-0.14975E 00	0.94630E	0.46254E-01	0.45733E	02	0.10682E	01	8-0.40566E 04
-0.14975E 00	0.94630E	0.46254E-01	0.45733E	02	0.10682E	01	9-0.40566E 04
-0.14975E 00	0.94630E	0.46254E-01	0.45733E	02	0.10682E	01	10-0.40566E 04
0.81320E-01	0.11422E	0.48235E-01	0.15754E	02	0.34800E	02	
0.81320E-01	0.11137E	0.48666E-01	0.15754E	02	0.34801E	02	1-0.62376E 04
0.81320E-01	0.11422E	0.48221E-01	0.15754E	02	0.34800E	02	2-0.59226E 04
0.81320E-01	0.11422E	0.48235E-01	0.15754E	02	0.34800E	02	3-0.60317E 04
0.81320E-01	0.11422E	0.48235E-01	0.15754E	02	0.34800E	02	4-0.60234E 04
0.81320E-01	0.11422E	0.48235E-01	0.15754E	02	0.34800E	02	5-0.60235E 04
0.81320E-01	0.11422E	0.48235E-01	0.15754E	02	0.34800E	02	6-0.60235E 04

Distribution of calibration radio sources in the sky

There are at least two possible approaches. One is to observe extensively a small number of bright radio sources with accurately known positions. Another is to observe a large number of sources with comparatively few observations per source.

The first approach is desirable insofar as less slewing time is required, the observing procedure is made simpler, and only a few bright and well known sources are needed. In an altazimuth instrument, where large excursions in both altitude and azimuth are possible while tracking a single source, the first approach may be practical. In an equatorial telescope, the need to obtain a good sampling in declination requires that a substantial number of radio sources be observed. The declination calibration of the 300 foot transit telescope presents in this respect an analogous problem.

Subject to further detailed selection, we could take for a starting point the list of sources contained in Mike Davis' memorandums of September 14 and October 25, 1971. These lists are reproduced below.

There is no simple way to plan in advance a detailed sequence of observations since almost certainly it will be necessary to adapt to circumstances of the moment. While observing, the declination and hour angles should be plotted and observing scheduled to make sure there are no large gaps in the coverage of the sky. This should be done separately during the day and night times, since we have reason to expect better pointing at night and it will be desirable to make separate day and night time analysis.

Calibrators

Source	Ref.	Source	Ref.
3C 8	CT	3C 288	RR
0019-00	RR	3C 289	PR
3C 12	CT	1345+12	RR
3C 20	C	1354+19	CT
3C 22	CT	3C 298	NRI
0106+01	NRI	3C 299	NRI
3C 55	C	3C 303	CT
0229+13	NRI	3C 305	C
3C 79	CT	3C 309.1	NRI
NRAO 140	NRI	3C 317	CT
CTA 26	CT	3C 318	RR
3C 94	CT	3C 319	C
0357-16	CT	3C 323.1	NRI
3C 137	CT	3C 324	RR
3C 153	RR	3C 330	CT
3C 166	C	3C 336	NRI
3C 171	CT	3C 341	CT
3C 173.1	CT	3C 338	C
3C 175.1	C	3C 343	NRI
3C 179	CT	3C 343.1	NRI
3C 180	CT	3C 345	NRI
3C 194	RR	1645+17	RR
3C 196	RR	3C 351	C
3C 200	RR	1732-09	CT
4C 39.25	RR	1756+13	CT
3C 231	CT	3C 371	NRI
3C 245	RR	3C 379.1	C
3C 249.1	NRI	3C 388	CT
3C 254	RR	3C 390	RR
1127-14	NRI	3C 401	RR
3C 263	NRI	3C 409	NRI
3C 265	C	3C 411	CT
3C 268.1	CT	3C 422	CT
3C 273	OPT	3C 427.1	NRI
1229-02	OPT	3C 429	CT
3C 275.1	RR	3C 433	CT
3C 277.1	RR	2127+04	NRI
3C 280	NRI	3C 438	CT
3C 287	NRI	2216-03	OPT
3C 286	NRI	3C 446	NRI
		CTA 102	NRI

C = Cambridge
 RR = Malvern
 CT = Caltech
 NRI = NRAO inter-ferometer
 OPT = Optical

Calibrators

3C 8	1229-02
3C 12	1237-10
3C 20	3C 275.1
3C 22	3C 277.1
0106+01	3C 280
3C 43	1306-09
3C 45	3C 287
3C 47	3C 288
3C 52	3C 289
3C 79	1345+12
0310-15	3C 295
4C 32.14	3C 298
CTA 26	3C 299
3C 93	3C 300
3C 93.1	3C 303
3C 94	3C 305
0357-16	3C 318
3C 131	3C 319
3C 132	1523+03
3C 137	3C 323.1
3C 166	3C 324
3C 171	3C 332
3C 173.1	1621-11
3C 175.1	3C 336
0735-17	3C 343
3C 194	3C 343.1
3C 196	1645+17
4C 39.25	3C 351
3C 231	1730-13
3C 244.1	1732-09
3C 245	3C 381
3C 254	3C 390
1127-14	3C 409
1138+01	3C 411
3C 263.1	3C 422
3C 265	3C 424
3C 267	3C 429
3C 268.1	3C 430
3C 268.3	3C 433
3C 273	2127+04
	2216-03

Optimum observing wavelength

A number of factors affect the choice of an operating wavelength for pointing calibration purposes. Let us define a performance index, to allow a quantitative comparison between wavelengths, as follows:

$$PI \propto \frac{1}{HPBW} \times \frac{1}{T \text{ system noise}} \times \text{Bandwidth}^{1/2} \times \text{Mean Source}$$

$$\text{Flux density} \times \text{Gain loss factor} \times \text{Beam distortion factor}$$

The following proportionalities hold for the individual factors

$$HPBW \propto \lambda$$

$$T \text{ system noise} \propto \text{cnt. for centimetric wavelengths}$$

(Provided receivers of comparable quality are available, about 150°K)

$$\text{Bandwidth} \propto \lambda^{-1}$$

$$\text{Flux density} \propto \lambda^{0.6}$$

(Average spectral index for centimetric wavelengths, Toth, Kellerman, Davis, External Galaxies and Quasi Stellar Objects, p. 444, 1972)

$$\text{Gain loss factor} \propto e^{-\left(\frac{\lambda_0}{\lambda}\right)^2}$$

where λ_0 represents a short wavelength limit of the telescope.

It is difficult to agree on particular numbers. 2 cm can be considered reasonable for the 140 foot telescope.

$$\text{Beam distortion} \propto e^{-\left(\frac{\lambda_0}{\lambda}\right)^2}$$

A beam distortion factor is required because a distorted beam shape will introduce systematic errors in the Gaussian fitting process. It is difficult to define precisely. Tentatively take the same expression as for the gain loss.

Combining all the above expressions we have

$$P.I \propto \lambda^{-0.9} e^{-2\left(\frac{\lambda_0}{\lambda}\right)^2}$$

Tabulating this expression for some wavelengths of interest, with an arbitrary proportionality factor:

<u>λ cm</u>	<u>PI</u>
21	6
11	11
6	16
3	15
2	7

The best wavelength appears to be 6 cm.

Example of least squares parameter fit

A program listing is reproduced next to provide an example of application of the previous theory to fit the standard pointing curves, also reproduced for reference.

Different fits have been tried and the one given as an example does not include all the terms previously discussed. The rms of the residuals is 5" in declination and 8" in hour angle, for points in the range 60° to -10°, and 4h to -4h in hour angle. 35 points, as indicated in the curves, were included. No detailed description of the program will be provided at this time since it is not definitive and representative of the final formats.

The rms errors are comparable to those obtained by Pauliny Toth, who used observational data and obtained, as should be expected, somewhat larger errors.

```
C PCCC
  DIMENSION C(1000),CA(50),CT(1000),CTC(1000),
  * WOKKA(50),WOKKR(50),
  X PES(100),
  X CLS(1000),
  X DFLA(2),DELT(50),PARM(50)
  REAL L
  READ 200,NO
200 FORMAT(I3)
  PRINT 200,NO
  RASE=4.85E-6
  L=35/57.3
  NP=13
  NRC=0
```

```
C
  DO 100 NCU=1,NO
  READ 210,D,H,DELD,DELH
210 FORMAT(4F10.5)
  PRINT 211,NCU,D,H,DELD,DELH
211 FORMAT(I5,4F10.5)
  D=D/57.3
  H=H*15/57.3
  DELD=DELD/(50*57.3)
  DELH=DELH*COS(D)/(240*57.3)
  CA(1)=1
  CA(2)=0
  CA(3)=0
  CA(4)=-COS(D)
  CA(5)=0
  CA(6)=-1
  CA(7)=0
  CA(8)=SIN(D)
  CA(9)=-SIN(H)
  CA(10)=COS(H)*SIN(D)
  CA(11)=COS(H)
  CA(12)=SIN(H)*SIN(D)
  CA(13)=(-COS(L)*COS(H)*SIN(D)+SIN(L)*COS(D))/
  X (COS(L)*COS(H)*COS(D)+SIN(L)*SIN(D))
  CA(14)=-COS(L)*SIN(H)/
  X (COS(L)*COS(H)*COS(D)+SIN(L)*SIN(D))
  CA(15)=-COS(L)*COS(H)*SIN(D)+SIN(L)*COS(D)
  CA(16)=C
  CA(17)=0
  CA(18)=-COS(L)*SIN(H)
  CA(19)=0
  CA(20)=SIN(D)*SIN(H)
  CA(21)=SIN(D)
  CA(22)=0
  CA(23)=COS(D)
  CA(24)=0
  CA(25)=0
  CA(26)=COS(H)
  IF(NRC.EQ.0) GO TO 150
  GO TO 160
150 CALL MCPY(CA,CT,2,NP,0)
  GO TO 161
160 CONTINUE
  CALL RTIE(C,CA,CT,NRC,NP ,0,C,2)
```



```
161 CONTINUE
   NRC=NRC+2
   CALL MCPY(CT,C,NRC,NP,0)
   DELT(NRC-1)=DELD
   DELT(NRC)=DELH
100 CONTINUE
```

C
C

```
   CALL GMTRA(C,CT,NRC,NP)
   CALL GMPRD(CT,C,CTC,NP,NRC,NP)
   CALL MINV(CTC,NP,DET,WORKA,WORKB)
   PRINT 112,DET
112 FORMAT(6E12.4)
   CALL GMPRD(CTC,CT,CLS,NP,NP,NRC)
   CALL GMPRD(CLS,DELT,CTC,NP,NRC,1)
```

C

```
   DO 110 I=1,NP
   CTCA=CTC(1)/4.85E-6
   PRINT 111,I,CTCA
111 FORMAT(13,F14.3)
110 CONTINUE
   CALL GMPRD(C,CTC,RES,NRC,NP,1)
   CALL GMSUB(DELT,RES,RES,NRC,1)
   RMSH=0
   RMSD=0
   DO 180 I=1,NU
   RMSD=RMSD+RES(2*I-1)**2
   RMSH=RMSH+RES(2*I)**2
180 CONTINUE
   RMSD=RMSD/NU
   RMSH=RMSH/NU
   RMSD=SQRT(RMSD)/4.85E-6
   RMSH=SQRT(RMSH)/4.85E-6
   DO 191 I=1,NU
   RESA=RES(2*I-1)/BASE
   RESB=RES(2*I)/BASE
   PRINT 192,I,RESA,RESB
192 FORMAT(15,2F12.3)
191 CONTINUE
   PRINT 190,RMSD,RMSH
190 FORMAT(2F12.3)
   STOP
   END
```

30*

1	60.00000	0.0	1.40000	5.20000
2	60.00000	-3.00000	2.10000	1.60000
3	40.00000	4.00000	1.90000	8.20000
4	40.00000	3.00000	1.60000	6.00000
5	40.00000	2.00000	1.40000	4.30000
6	40.00000	1.00000	1.40000	3.00000
7	40.00000	0.0	1.40000	2.00000
8	40.00000	-1.00000	1.60000	1.30000
9	40.00000	-2.00000	1.80000	0.50000
10	40.00000	-3.00000	2.20000	-0.10000
11	40.00000	-4.00000	2.50000	-1.40000
12	30.00000	3.00000	1.50000	4.20000
13	30.00000	0.0	1.30000	1.00000
14	30.00000	-3.00000	2.20000	-1.00000
15	15.00000	4.00000	2.30000	6.20000
16	15.00000	3.00000	1.60000	-3.90000
17	15.00000	2.00000	1.30000	2.30000
18	15.00000	1.00000	1.10000	1.50000
19	15.00000	0.0	1.10000	0.50000
20	15.00000	-1.00000	1.20000	0.0
21	15.00000	-2.00000	1.50000	-0.80000
22	15.00000	-3.00000	2.00000	-1.90000
23	15.00000	-4.00000	2.50000	-3.50000
24	0.0	3.00000	1.50000	4.50000
25	0.0	2.00000	1.20000	3.00000
26	0.0	0.0	1.00000	0.40000
27	0.0	-2.00000	1.40000	-2.50000
28	0.0	-3.00000	2.10000	-5.40000
29	-10.00000	3.00000	1.60000	3.80000
30	-10.00000	2.00000	1.20000	2.30000
31	-10.00000	1.00000	1.00000	1.60000
32	-10.00000	0.0	1.00000	0.50000
33	-10.00000	-1.00000	1.20000	-0.30000
34	-10.00000	-2.00000	1.50000	-1.90000
35	-10.00000	-3.00000	2.00000	-3.50000

0.4211E 03

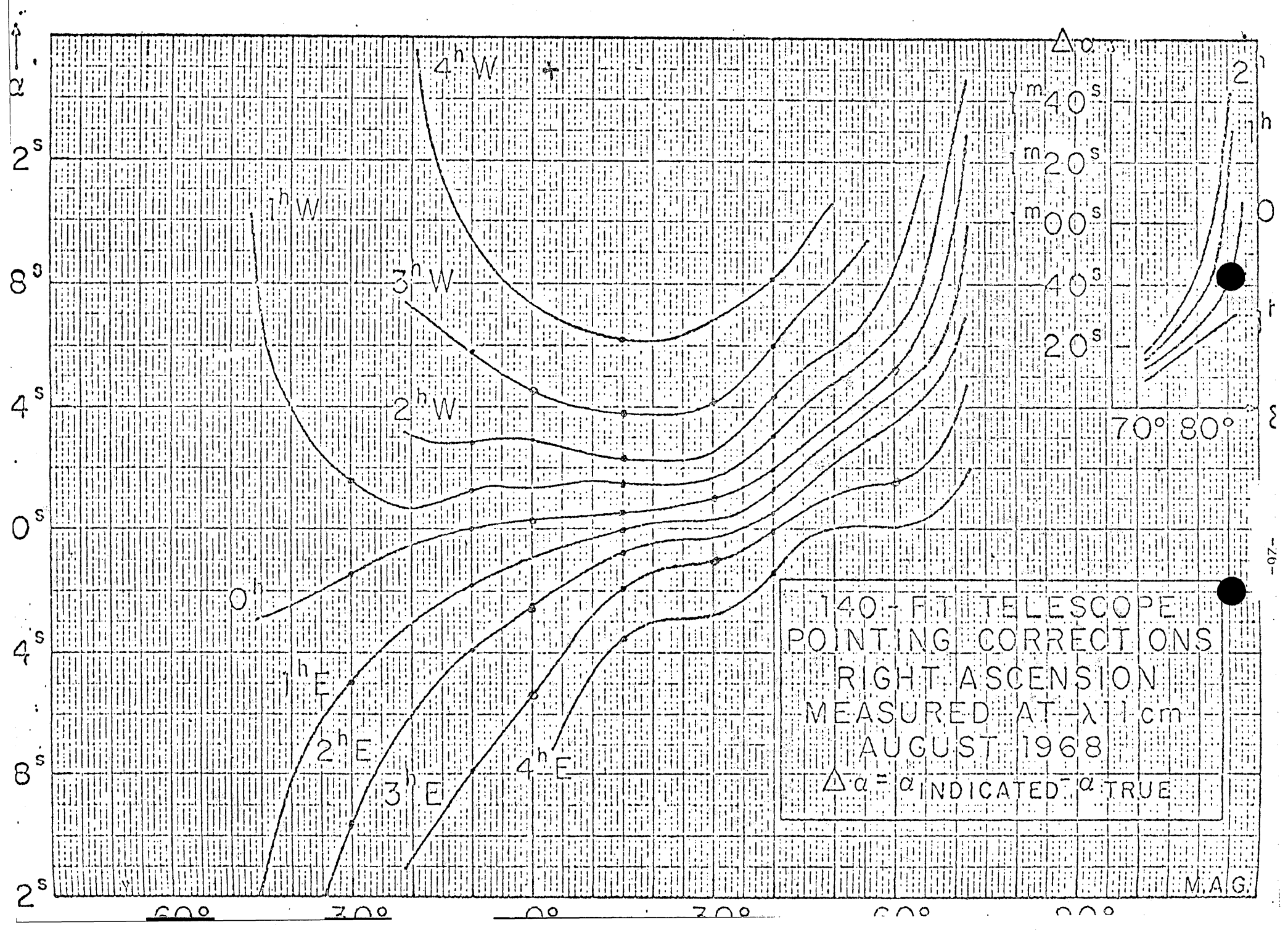
1	302.616
2	52.239
3	-77.283
4	1.883
5	13.066
6	-199.422
7	-31.443
8	-155.312
9	-39.875
10	185.535
11	-127.181
12	50.472
13	-23.980

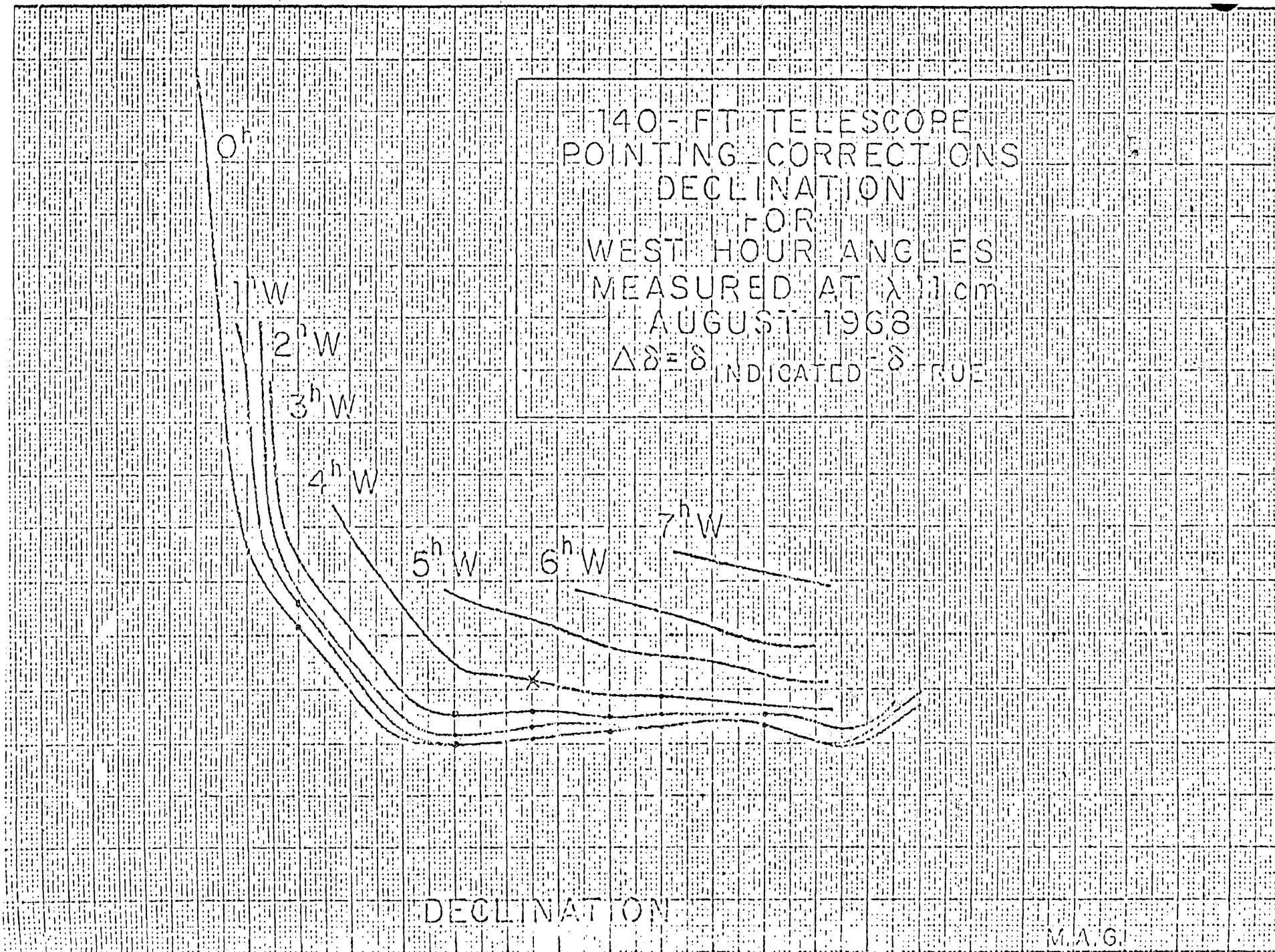
1	-8.335	-5.477
2	-3.763	-0.214
3	-3.229	11.423
4	-2.850	2.245
5	-1.220	-2.764
6	5.672	-4.229
7	4.881	-3.142
8	8.316	0.548
9	4.705	-2.355
10	7.586	6.049
11	1.465	2.087
12	50.317	-7.407
13	3.813	-5.058

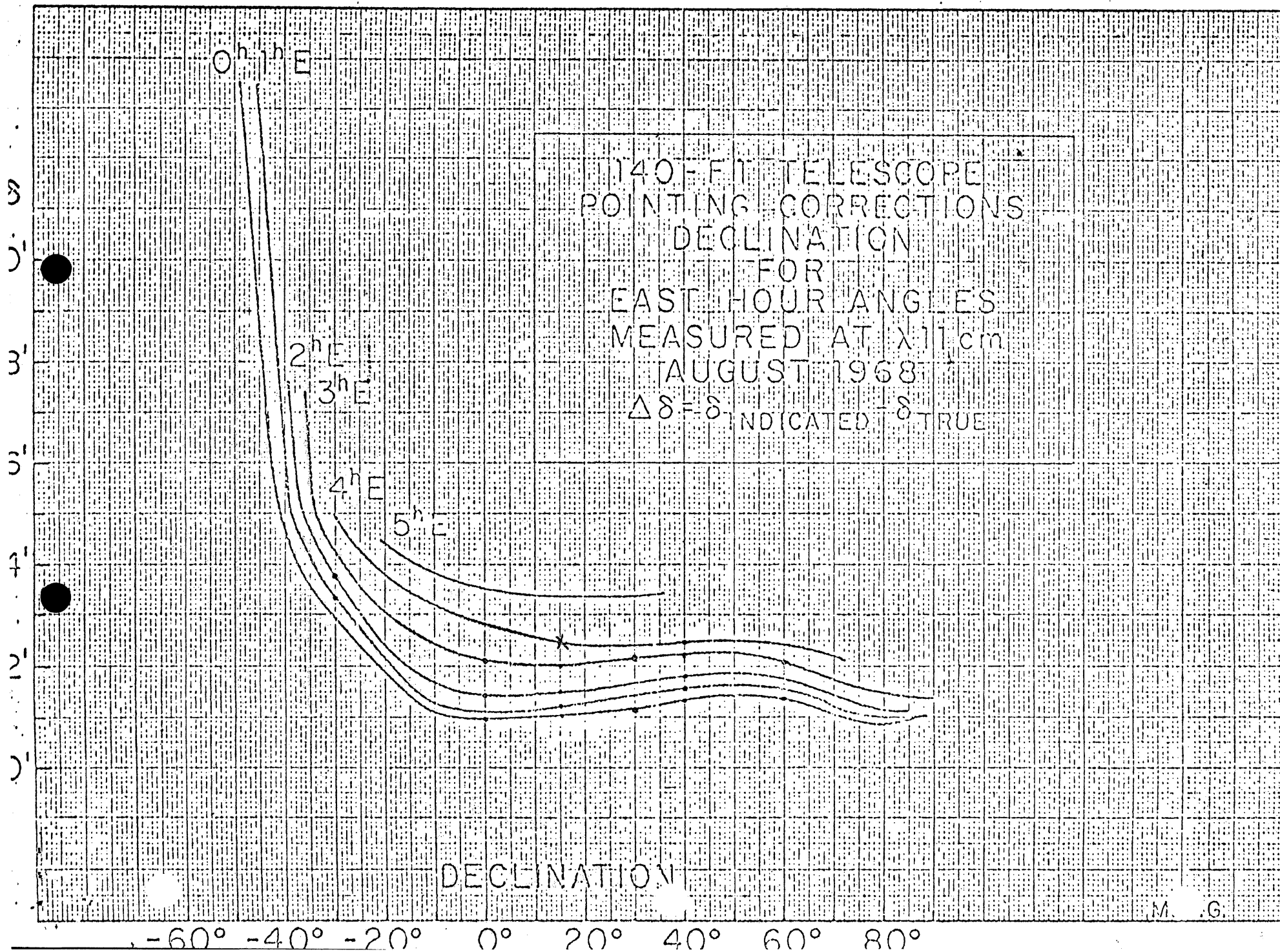
10	2.902	2.902
17	-1.196	6.626
18	-2.390	1.440
19	-1.821	-0.771
20	-5.744	5.823
21	-7.266	-7.152
22	-4.240	4.186
23	-4.626	-3.297
24	-7.740	4.837
25	-3.498	6.212
26	-1.217	4.929
27	-9.568	-7.278
28	2.696	-34.514
29	2.582	-14.327
30	1.230	-7.540
31	3.067	5.191
32	4.785	3.985
33	5.711	7.978
34	1.157	9.386
35	1.023	6.079

5.519

8.368







Structural and thermal focal point changes

Focusing is only indirectly related to the pointing calibration, but an understanding of the causes of focal point changes is nevertheless desirable to obtain observations in minimum time and with the highest quality. Eventually, an automatic focusing mode is desirable and in that respect it is important to be able to estimate the best focusing, as accurately as possible.

Gravitational deformations and thermal deformations are the main causes of focus changes. If the gravitational loads are decomposed along the three axis of the astrometric system, it is possible to determine that the Y and Z loadings, anti-symmetric with respect to the planes of symmetry, will produce no changes in focal point position. This assumes that reactions at the declination drive are small, so that the asymmetry introduced is negligible (this assumption is normally well justified). The x component of the gravitational load does produce focal changes of the form

$$\Delta f = A_n \cos z$$

where A_n is a constant and z is the zenith distance. Referring to the section on refraction, we can express the $\cos z$ and substitute as follows

$$\Delta f = A_n (cL cH cD + sL sD)$$

Typically gravitational changes will be of the order of 1 inch. Thermal effects may be quite important. No detailed computations have been carried out for the 140' telescope but representative values can be extracted from 65 meter telescope design (von Hoerner and Herrero, 65 Meter Report No. 37, February 20, 1971).

Focal point motion can be as large as 1 mm per °F of axial temperature gradient across the reflector structure (this thermal mode is the most important). With temperature gradients of up to 10°F in the daytime, changes of 10 mm may occur.

To correct for these effects, measurements with temperature sensors in the telescope structure are necessary. This is a routine procedure with the 36 ft. telescope in Tucson. Present temperature sensors at the 140' telescope were installed with a statistical analysis of temperature differences in mind, and although not optimally located for gradient measurements, will be useful.

Mezger et al, August 1966 report on 140' tests, gives some curves of focal length changes as a function of the coordinates, (Fig. 10). Their usefulness is very limited because no thermal measurements were taken simultaneously.

Thermal focus changes can be set proportional to the observed temperature gradient across the thickness of the backup structure, neglecting other small thermal effects.